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Citation for final published version:

Mueller, Stephanie, Muhawenimana, Valentine , Wilson, Catherine and Ouro, Pablo 2021. Experimental investigation of the wake characteristics behind twin vertical axis turbines. *Energy Conversion and Management* 247 , 114768. 10.1016/j.enconman.2021.114768

Publishers page: <https://doi.org/10.1016/j.enconman.2021.114768>

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Experimental investigation of the wake characteristics behind twin vertical axis turbines

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Abstract

Vertical axis wind and tidal turbines are a promising technology, well suited to harness kinetic energy from highly turbulent environments such as urban areas or rivers. The power density per occupied land area of two or three vertical axis rotors deployed in close proximity can notably exceed that of their horizontal axis counterparts. Using acoustic Doppler velocimetry, the three-dimensional wake developed downstream of standalone and twin vertical axis turbines of various shaft-to-shaft distances and rotational direction combinations was characterised in terms of mean velocity and turbulence statistics, with their impact on momentum recovery quantified. Results show that the wake hydrodynamics were more impacted by turbine rotational direction than lateral distance between devices for the range of lateral spacing considered. In the cases with turbines operating in a counter-rotating forward configuration, the wake mostly expanded laterally and attained the largest velocities that exceeded those in the single turbine case, with full momentum recovery at 5 turbine diameters downstream. The wake developed by the counter-rotating backward setup notably extended over the vertical direction, whilst devices rotating in the same direction featured the greatest lateral wake expansion with reduced velocities. Linear wake superposition of the single turbine wake provided a good representation of the mean velocity field behind twin-turbine setups. The presented results indicate that, in the design of twin-turbine arrays moving in counter-rotating forward direction, a lateral spacing of, at least, two turbine diameters should be kept as this allows the kinetic energy in the wake to be fully recovered by five turbine diameters downstream.

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Keywords: vertical axis turbine, VAWT, wakes, river turbine, turbine array, turbulence

1. Introduction

The continuous increase in energy demand and current change in energy policy towards net-zero carbon economies is leading to the rapid expansion and development of new sustainable renewable energy technologies. To date, hydropower is the second-largest source of renewable energy [1] and has long been considered as an environmentally friendly and clean form of energy generation. However, the growing concern about the ecological impact of traditional large-scale hydropower projects [2] is propelling the development of small-scale, low-head hydro-kinetic alternatives [3], e.g. horizontal (HAT) and vertical (VAT) axis river turbines.

There has been an increased interest by the research community to improve the design of VATs, both to enhance the rotor's efficiency and deploy VATs in arrays to take advantage of flow acceleration due to blockage effects of closely located turbine rotors [4] that increase by 10 times the installed power capacity of HATs [5]. VATs offer a wide range of mechanical and hydrodynamic advantages compared to HATs. Their vertical rotational axis, for instance, allows positioning of the generator and other heavy components on the ground or a floating platform, which diminishes their technical complexity compared to HATs and improves their suitability to river applications [6]. From an operational point of view, the relatively low rotational speed and rectangular cross-section that maximise the swept rotor area in constrained shallow waters make them particularly suitable for rivers and estuaries with low-to-medium flow velocities [6]. VATs operate independently of the flow direction, i.e. they are omnidirectional; hence, no yaw-angle correction and alignment with the flow direction is needed. Subsequently, these unique operational characteristics also have the advantage of potentially reducing the environmental impact by operating at lower rotational speed than HATs. This, in turn, lowers acoustic contamination [6] and presents a potential reason for the lack of injuries and mortality observed for VATs [7]. Despite this promising catalogue of benefits, VAT's main drawback remains their lower performance compared to HATs, however this can be overcome if several VAT rotors are deployed in close proximity as an array [8].

The wake developed by a single VAT has been extensively studied through small-scale experimental testing in open channels (such as those from Brochier et al. [9], Bachant and Wosnik [10], Araya et al. [11], Ouro et al. [12] or Strom et al. [13]), wind tunnels (e.g. studies from Tescione et al. [14], Kadum et al. [15], Rolin and Porté-Agel [16] or Vergaerde et al. [17]), and high-fidelity numerical simulations

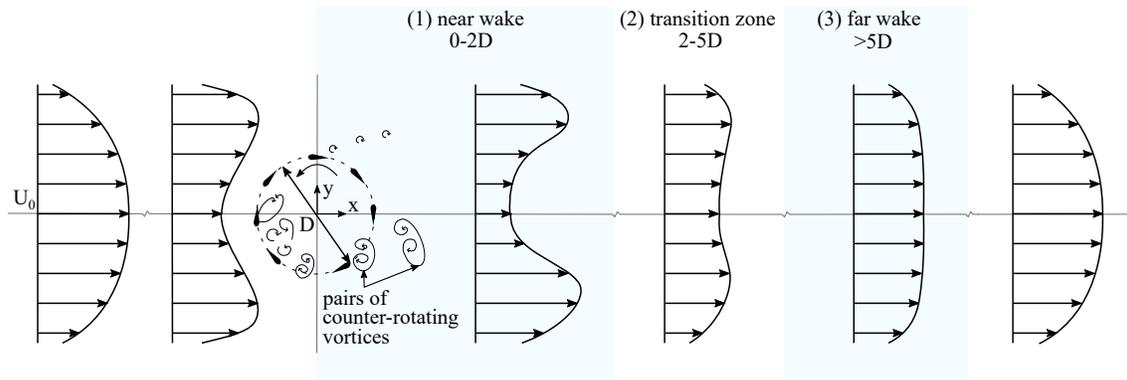


Figure 1: Wake evolution of a single, counter-clockwise rotating VAT of diameter D , consisting of three distinct regions: (1) a near-wake region ($x/D \leq 2$), characterised by a low-momentum region laterally bounded by shear layers that result from the advection of two pairs of counter-rotating vortices over the downstroke side and smaller vortices over the upstroke side; (2) a transition zone ($2 \leq x/D \leq 5$); and (3) a far-wake region ($x/D \geq 5$) characterised by wake recovery.

35 (including the work from Shamsoddin and Porté-Agel [18], Lam and Peng [19], Posa
 36 et al. [20], Abkar and Dabiri [21], Ouro and Stoesser [22], or Posa [23]). Only a
 37 handful of full-scale devices have been tested in field campaigns, e.g. in a wind
 38 turbine array [8] or in a tidal flow [24].

39 Based on the observations from these studies, the primary regions developed in
 40 the wake of a VAT in an open-channel are depicted in the schematic presented in
 41 Figure 1, which outlines three distinct regions, namely the near-wake ($x/D \leq 2$),
 42 transition zone ($2 \leq x/D \leq 5$), and far-wake ($x/D \geq 5$) [12], with x indicating
 43 the streamwise location and D being the turbine diameter. The near-wake region is
 44 characterised by the turbine-induced flow structures such as the two counter-rotating
 45 vortices shed by the blades when undergoing dynamic stall during the downstroke
 46 phase [9]. The latter coherent turbulent structures generate a shear layer that isolates
 47 the low-momentum region developed in the near-wake core from the high-velocity
 48 region outside the wake, thus limiting entrainment of the surrounding flow [12]. The
 49 blades experience lower flow separation over the upstroke rotation as their relative
 50 velocity is larger than during their downstroke motion, which also prevents deep
 51 dynamic stall [22]. The shedding pattern of these turbulent structures depends on
 52 the tip-speed ratio, i.e., the relative blade velocity to that of the approaching flow.
 53 This unevenly generated flow during the downstroke and upstroke motion of the
 54 rotor blades can render the near-wake asymmetric about its centreline [14]. Within
 55 the transition zone, the wake starts to vertically and laterally expand with a larger

56 ambient turbulent flow entrainment that increases the turbulent fluxes and intensity,
 57 and momentum begins to recover at a faster rate [12]. In the far-wake region, the core
 58 momentum further recovers with increasing downstream distance until it eventually
 59 reaches a mean velocity value similar to that of the free stream flow at distances that
 60 vary with the turbine’s aspect ratio [25] and dynamic solidity [11].

61 2. Problem statement

62 To unfold the full potential of VATs, there remains a need for a detailed under-
 63 standing of wake hydrodynamics of multiple VATs in order to identify their optimal
 64 arrangement and thus maximise the harnessed kinetic energy when deployed in ar-
 65 rays [26]. The pilot wind-energy project FLOWE [8], for instance, showed that VATs
 66 could achieve a higher power density than HATs when deployed in twin configura-
 67 tions. To date, VAT wake interactions have been studied mostly for side-by-side
 68 twin-turbine setups, mainly focusing on the turbines’ rotational direction with nu-
 69 merical simulations [27] and wind tunnel testing [17] and less on the shaft-to-shaft
 70 lateral spacing and relative alignment to the incident flow direction [28].

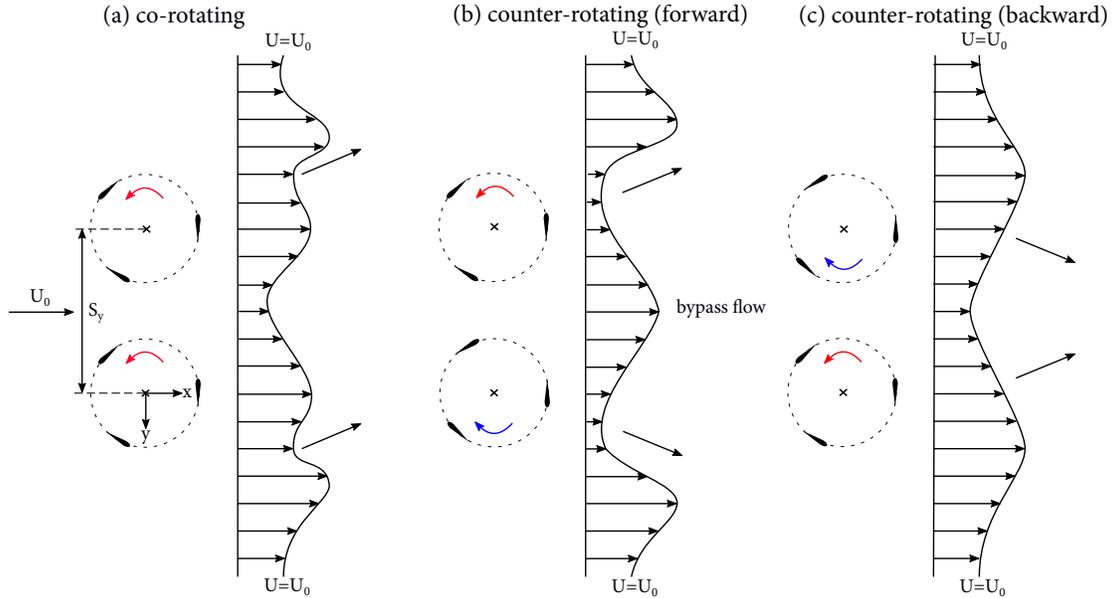


Figure 2: Wake interaction of three twin-VAT arrangements varying in rotational direction, laterally spaced by S_y , namely (a) co-rotating, (b) counter-rotating forwards, and (c) counter-rotating backwards.

71 Figure 2 depicts twin-VAT setups with devices co-rotating or counter-rotating
72 forward and backward, corresponding to cases in which blades move with or against
73 the flow direction in the bypass region. Comparison of the wake evolution for these
74 configurations shows that the individual wakes of co-rotating VATs (Figure 2a) evolve
75 independently in the downstream direction, with a reduced interference in the near-
76 wake [29]. These wakes start to interact with each other and partially merge at a
77 downstream distance that depends on the relative shaft-to-shaft separation (S_y/D)
78 [27]. The individual wakes of two turbines in the counter-rotating forward configu-
79 ration (Figure 2b), on the other hand, spread outwards in an axisymmetric fashion,
80 leading to laterally expanded wakes that progressively diverge with increasing down-
81 stream distance and creating high momentum flow region between both turbines.
82 Conversely, in the counter-rotating backward case, a prolonged combined wake is ob-
83 served after the transition zone. Both individual wakes progressively move towards
84 each other before merging and interacting [29]. This results in a lower high-velocity
85 bypass region [27] that varies with both turbine rotational direction and intra-turbine
86 spacing, i.e. smaller S_y/D values result in a higher flow blockage within the bypass
87 region that reduces the flow velocity in this area.

88 The wake patterns observed for twin-VATs have been mostly characterised in the
89 horizontal plane by two-dimensional simulations to study rotor position [30] and in-
90 termediate deflector influence [31] with limited experimental and numerical in-depth
91 studies looking at the three-dimensional wake evolution for multi-turbine arrange-
92 ments. In this paper, the three-dimensional wake hydrodynamics behind a single
93 turbine, and co- and counter-rotating twin-turbine setups are experimentally inves-
94 tigated using acoustic Doppler velocimetry measurements, and the impact of six
95 different lateral spacing and rotational direction combinations on the wake charac-
96 teristics are quantified. The paper is structured as follows: Section 3 describes the
97 experimental facility and techniques used, along with the turbine design and con-
98 figurations tested. Section 4 presents the mean velocity and turbulence statistics,
99 cross-averaged values of velocity and turbulence intensity over the measured wake
100 length for the single turbine and twin-VAT cases, and linear and quadratic wake
101 superposition is applied to predict twin-VAT wake dynamics. Conclusions are drawn
102 in Section 5.

103 3. Methods

104 Details of the experimental facility in which the tests were performed, turbine
105 rotor dimensions and operation, and the flow measurement method are provided in
106 this section.

107 *3.1. Flume setup*

108 The experiments were conducted in the Hydro-environmental Research Centre’s
 109 hydraulic laboratory at Cardiff University, UK. The experimental setup, depicted in
 110 Figure 3, used a 10m long, 1.2m wide, and 0.3m deep recirculating flume with a slope
 111 of 0.001. Flow depth and discharge were controlled by a pump and a tailgate weir
 112 which were located at the downstream end of the flume and kept constant throughout
 113 the experiment. Flow depth was measured using a Vernier pointer gauge with an
 114 accuracy of $\pm 0.1\text{mm}$ while discharge was measured with an ultrasonic flowmeter
 115 (TecFluid Nixon CU100) with a precision of $\pm 1.5\%$. Prior to rotating the VAT in each
 116 of the tests, sub-critical uniform flow with a discharge of $Q = 0.053\text{m}^3\text{s}^{-1}$ and a flow
 117 depth of $h_0 = 0.23\text{m}$ were established. Further hydraulic parameters are presented
 118 in Table 1, including cross-section averaged bulk velocity $U_0 = Q/A$, bulk Reynolds
 119 number ($Re = U_0 R_H / \nu$ with ν denoting the fluid kinetic viscosity and R_H the
 120 hydraulic radius), Reynolds number based on turbine diameter D ($Re_D = U_0 D / \nu$),
 121 and Froude number ($Fr = U_0 / \sqrt{gh_0}$).

Table 1: Details of hydraulic parameters adopted in the experiments, including flow discharge (Q), water depth (h_0), bulk velocity (U_0), bulk Reynolds number (Re), Reynolds number based on the turbine’s rotor diameter (Re_D), and Froude number (Fr).

Q	h_0	U_0	Re	Re_D	Fr
$[\text{m}^3\text{s}^{-1}]$	$[\text{m}]$	$[\text{ms}^{-1}]$	$[-]$	$[-]$	$[-]$
0.0053	0.23	0.19	$3.16 \cdot 10^4$	$2.28 \cdot 10^4$	0.13

122 *3.2. Description of the VATs*

123 The adopted VATs were manufactured with a rotor diameter $D = 0.12\text{m}$ and
 124 height $H = 0.12\text{m}$, i.e. with an aspect ratio H/D equal to unity. The rotor comprised
 125 three blades ($N_b = 3$) that were 3D printed with laser-sintered PA 2200 material
 126 conforming to a NACA 0015 airfoil profile geometry with zero preset pitch angle
 127 and 0.03m chord length (c), which yielded a geometric solidity $\sigma = N_b c / \pi D \approx 0.24$.
 128 DC motors (Nider DMN37K50G18A, DC 12V) were used in each turbine to impose
 129 a constant rotational speed $\Omega = 59\text{rpm}$ for an optimum tip-speed ratio $\lambda = 1.9$
 130 [12]. Each blade was attached to a main circular shaft of 0.006m diameter using
 131 two horizontal struts of 0.003m diameter, attached at vertical positions 0.01m away
 132 from the bottom and top tips of the blades; both components were made of stainless
 133 steel. The bottom end of the turbine shaft was connected to a bearing attached to
 134 the flume bed, leaving a clearance of 0.02m to the bottom tip of the blades. The

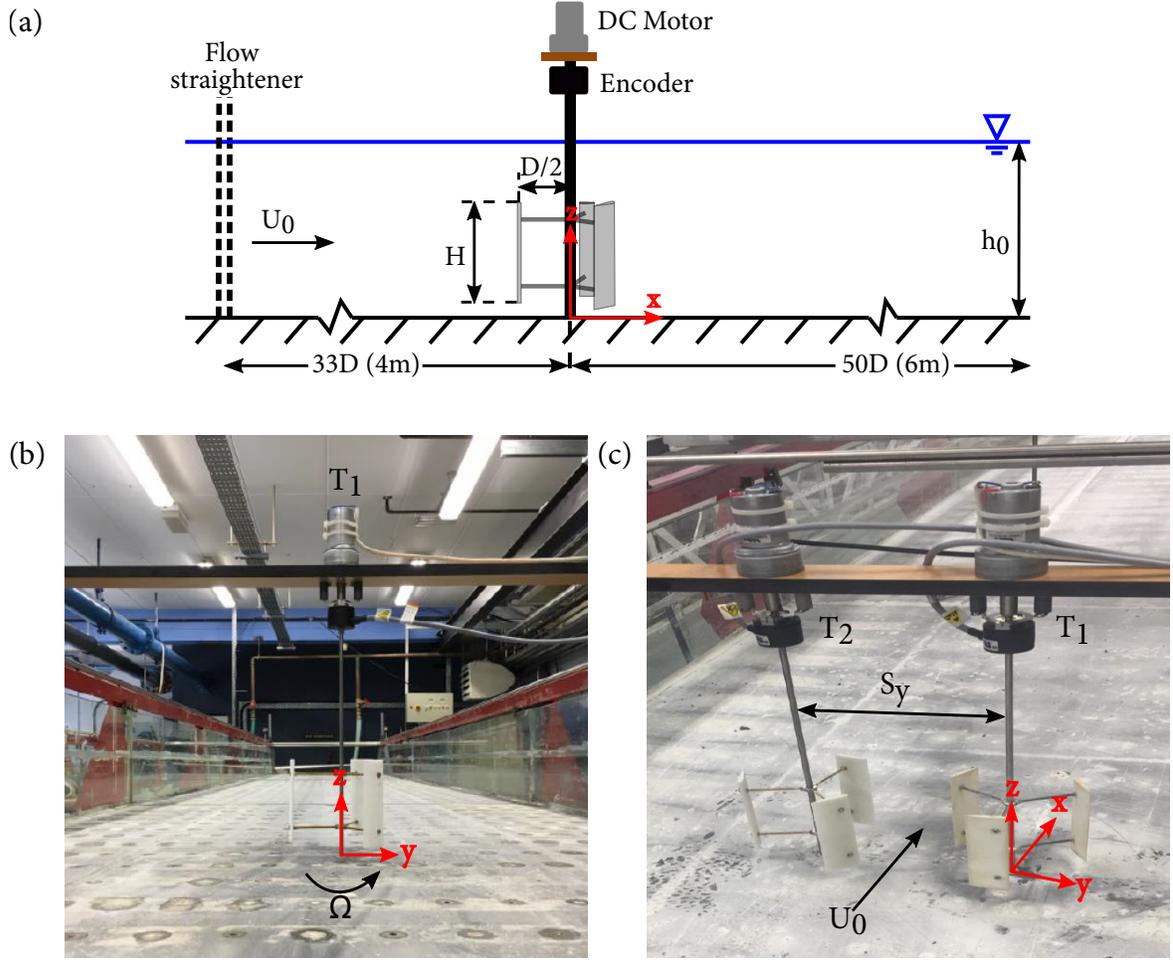


Figure 3: (a) Schematic of the experimental setup, depicting the streamwise location of a single VAT of height H and radius $D/2$, (b) photograph of the VAT T_1 located at the flume centre, $33D$ downstream of the flume inlet and rotating in counter-clockwise direction, (c) photograph of twin-VAT setup comprising of VATs T_1 and T_2 laterally spaced by a distance of S_y .

135 upper end of the shaft was connected to an encoder (Kübler, 5-30VDC, 100mA) that
 136 measured the rotational speed.

137 The first turbine, T_1 , was placed at the centre of the flume cross-section as shown
 138 in Figure 3 at a distance of 4m downstream of the flume inlet. The position of T_1 was
 139 set as the coordinate origins, considering as positive x -coordinates the streamwise
 140 flow direction, the positive lateral (y) direction over the right-hand side of the flume,
 141 and the z -coordinates in the upward direction starting at the flume's bed. In the array

Table 2: Details of the single and twin-turbine configurations test cases, including rotational direction and lateral spacing between turbines (S_y).

Test case	Description	Direction of rotation	S_y/D
ST	Single turbine	T_1 anti-clockwise	-
SR-1.5	Same rotation	T_1 and T_2 anti-clockwise	1.5
CRF-1.5	Counter-rotating forward	T_1 clockwise and T_2 anticlockwise	1.5
CRB-1.5	Counter-rotating backward	T_1 anti-clockwise and T_2 clockwise	1.5
SR-2.0	Same rotation	T_1 and T_2 anti-clockwise	2.0
CRF-2.0	Counter-rotating forward	T_1 clockwise and T_2 anticlockwise	2.0
CRB-2.0	Counter-rotating backward	T_1 anti-clockwise and T_2 clockwise	2.0

142 configurations, the second turbine T_2 was placed at the same streamwise location with
 143 a lateral shaft-to-shaft separation of S_y , as indicated in Figure 4.

144 A total of six twin-turbine configurations were tested, whose details, including
 145 turbine rotational direction and intra-spacing, are summarised in Table 2. First,
 146 a single, counter-clockwise rotating turbine (Figure 3b) was tested to characterise
 147 an individual VAT wake to be used as a reference wake distribution for the twin-
 148 turbine wake analysis. In the twin configurations, two shaft-to-shaft intra-turbine
 149 spacings, S_y of $1.5D$ and $2.0D$, were considered, in combination with the different
 150 rotational directions for each turbine, as depicted in Figure 2. The co-rotating case
 151 (hereinafter denoted as "same rotation" SR setup) considered both VATs to move
 152 with counter-clockwise motion (Figure 2 (a)). Then, two scenarios with turbines
 153 rotating in opposite directions, namely counter-rotating setups, were tested. These
 154 include the (1) "counter-rotating forward" CRF setup (Figure 2 (b)) in which the
 155 blades moved in the flow direction in-between rotors (i.e., in the bypass flow region);
 156 and (2) the "counter-rotating backward" CRB setup in which blades travelled against
 157 the incident flow in this region (Figure 2 (c)). For instance, the setup CRF-1.5
 158 denotes the counter-rotating forward layout with an inter-turbine separation of $1.5D$.

159 3.3. Hydrodynamic measurements

160 Hydrodynamic measurements were conducted using a side-looking Acoustic Doppler
 161 Velocimeter (ADV) (Nortek Vectrino). To ensure sufficient data quality and capture
 162 of a representative sample of the high-frequency turbulence fluctuations character-
 163 istic from VAT wakes, sampling periods of 300s (cross-sections at $x/D = 1.0, 1.5$
 164 and 2.0) and 180s (cross-sections at $x/D > 2.0$) were adopted with a frequency of
 165 200Hz. Signal quality was enhanced by seeding the water with Spherical®110P8
 166 hollow glass spheres (Potters Industries LLC) with a mean particle size $11.7\mu\text{m}$ and
 167 specific gravity of 1.10g/cc .

168 To characterise the approach flow conditions, one lateral cross-section was mea-
 169 sured at 1m upstream of the turbine (approx. $8D$). This cross-section comprised
 170 six vertical velocity profiles laterally spaced by 0.1m, starting at $y/D = 0$. Each
 171 velocity profile consisted of 20 measurement points vertically spaced by $\Delta z = 0.01m$
 172 ($0.08D$), starting at 0.01m above the flume bed until approx. 0.03m below the wa-
 173 ter surface. Then, for each of the single and twin-turbine configurations, lateral
 174 cross-sections ($y - z$ planes) were measured at nine streamwise locations starting
 175 at $1D$ and reaching until $10D$ downstream of the turbine, as depicted in Figure 4
 176 (a). Each cross-section comprised 12 to 14 vertical velocity profiles in the lateral
 177 direction for the single and twin-turbine setups, respectively. Lateral spacing be-
 178 tween vertical profiles was 0.05m ($0.42D$) (ST: $-1.25 \leq y/D \leq 1.25$, twin-VATs:
 179 $-1.25 \leq y/D \leq 7.92$), and increased to 0.1m ($0.83D$) within the free-stream region,
 180 as shown in Figure 4 (b).

181 Velocity data were filtered and post-processed using Matlab (2019a). Data with
 182 SNR $\leq 15\%$ and COR $\leq 70dB$ were removed from the data set, after which data
 183 were despiked using an open-source toolbox [32] provided in [33]. The instantaneous
 184 filtered velocity vector $\mathbf{u} = (u, v, w)$ record was then divided using the Reynolds
 185 decomposition: $\mathbf{u}(t) = \bar{\mathbf{u}} + \mathbf{u}'(t)$, with the time-averaged operation denoted as $\overline{(\cdot)}$
 186 and the fluctuating components represented as $(\cdot)'$. Normalised turbulence statistics
 187 were computed in terms of streamwise turbulence intensity ($u'/U_0 = \sqrt{\overline{u'u'}}/U_0$),
 188 turbulent kinetic energy ($tke = 0.5(\overline{u'u'} + \overline{v'v'} + \overline{w'w'})/U_0^2$), and Reynolds shear
 189 stresses ($\overline{u'v'}/U_0^2$ and $\overline{u'w'}/U_0^2$). Cross-sectional plots are presented normal to the
 190 flow and looking in the downstream flow direction.

191 4. Results and Discussion

192 In this section, the ADV measurements that characterise the approach flow and
 193 the wake behind the single and twin-turbine configurations are presented. Further
 194 analysis is provided in terms of downstream evolution of the spatially-averaged wake
 195 velocity and wake superposition techniques.

196 4.1. Approach flow

197 The vertical distribution of normalised streamwise mean velocity (\bar{u}/U_0), stream-
 198 wise turbulence intensity (u'/U_0), and vertical Reynolds shear stresses ($\overline{u'w'}/U_0^2$) are
 199 presented in Figure 5 (a-c), respectively, for six lateral locations over the left half
 200 of the cross-section (the approach flow is deemed symmetric). The mean velocity
 201 profiles show a power-law distribution over the flow depth, with a nearly constant
 202 turbulence intensity distribution yielding a depth-averaged value of $u'/U_0 = 0.14$.

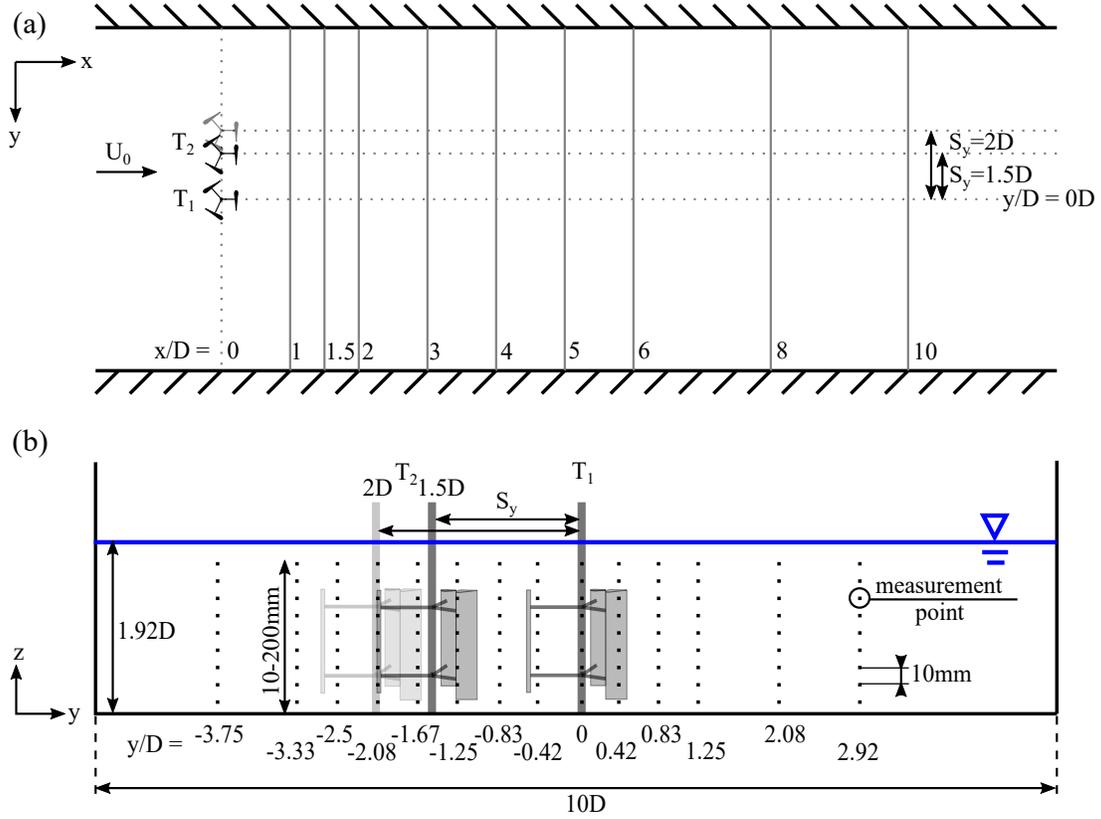


Figure 4: ADV wake measurement locations, showing (a) locations of cross-sections ($y - z$ plane) measured in streamwise direction, starting at $1D$ and until $10D$ downstream of the turbines, and (b) lateral distribution of the vertical measurement profiles over the flume section.

203 There is some non-uniformity in the $u'w'$ distribution between lateral locations, especially at $y/D = -4.17$, attributed to its proximity to the flume wall (located $0.83D$ away from the lateral wall), which can impact the distribution of turbulent fluxes [34].

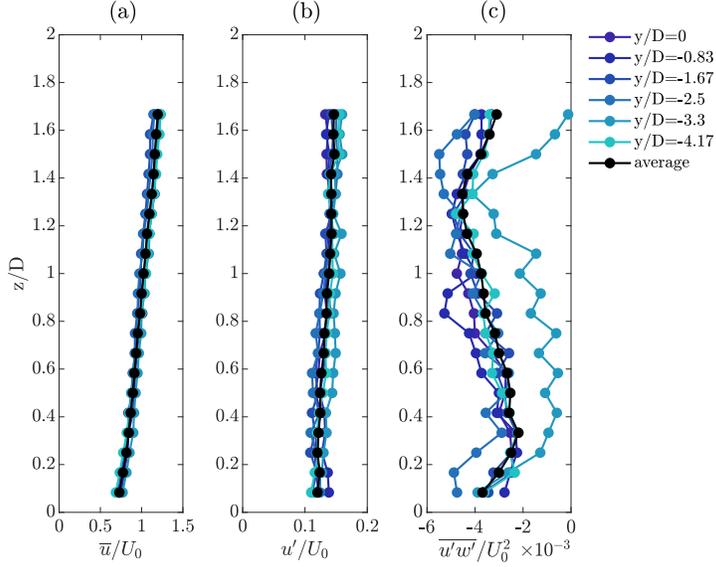


Figure 5: Upstream profiles measured at six lateral locations beginning at the flume centreline ($y/D = 0$). (a) Time-averaged streamwise velocity \bar{u} normalised by the bulk velocity U_0 , (b) streamwise turbulence intensity u' normalised by U_0 , and (c) vertical Reynolds shear stress $\overline{u'w'}$ normalised by U_0^2 . The black line and symbols correspond to the average value from the vertical measurements.

207 *4.2. Single turbine wake evolution*

208 Before examining the dynamics of twin-turbine wakes, the wake characteristics
 209 of the single turbine (ST) case are presented in $y - z$ -planes at downstream dis-
 210 tances (x/D) from one to five diameters to identify its key characteristics and three-
 211 dimensional evolution. The black rectangular outline in the contour plots represents
 212 the projected area of the turbine's rotor (see Figure 6).

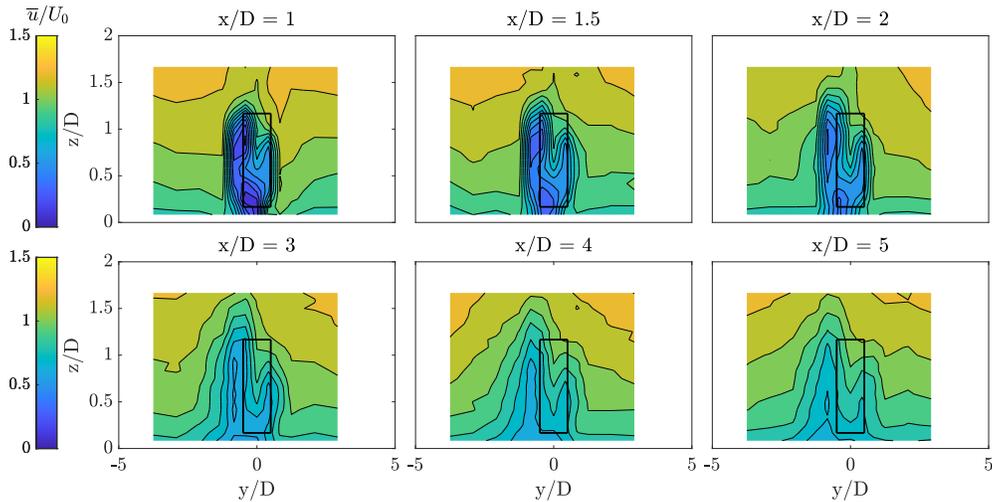


Figure 6: Contours of \bar{u}/U_0 at downstream cross sections located at $x/D = 1, 1.5, 2, 3, 4$ and 5 for the case with the single turbine (ST) rotating with anti-clockwise motion. The solid black rectangle represents the perimeter of the turbine’s rotor. Flume’s centreline is at $y/D = 0$.

213 The distribution of the normalised mean streamwise velocity (\bar{u}/U_0) is shown in
 214 Figure 6. The near wake ($x/D \leq 2$) behind the ST was characterised by a region of
 215 large velocity deficit immediately downstream of the rotor, which was particularly
 216 pronounced on the upstroke part ($y/D \leq 0$) of the blades’ rotation, i.e., when the
 217 blades move against the flow thus generating the highest relative velocity. This causes
 218 the wake to be asymmetric relative to the rotor centreline which has previously been
 219 observed by [9]. Until $x/D = 2$, the areas near the top and bottom tips in the
 220 upstroke side appeared to attain the minimum velocity values, likely arising from
 221 tip-vortices generated by the blades, similar to the PIV results presented in [16].
 222 Over the downstroke side ($y/D \geq 0$) the velocity was larger, with the lowest values
 223 distributed over the mid turbine height ($0.3 \leq z/D \leq 0.9$) rather than the tip
 224 location. Beyond $x/D = 3$, the transitional-wake region [12] was characterised by a
 225 vertical and lateral expansion of the low-velocity wake. In the far wake ($x/D \geq 6$,
 226 not shown here for brevity) most of the momentum was recovered, with velocities
 227 yielding values close to the approach flow velocity, but remnants of the wake signature
 228 were still visible over the whole water column. Overall, the wake evolution observed
 229 for the ST case is similar to the wake previously outlined in Figure 1, and those
 230 presented in [16] whose Reynolds number was almost one order of magnitude higher.

231 The distribution of turbulent kinetic energy (tke) is presented in Figure 7. Similar
 232 to the distribution of \bar{u}/U_0 , the upstroke side over the whole wake length featured

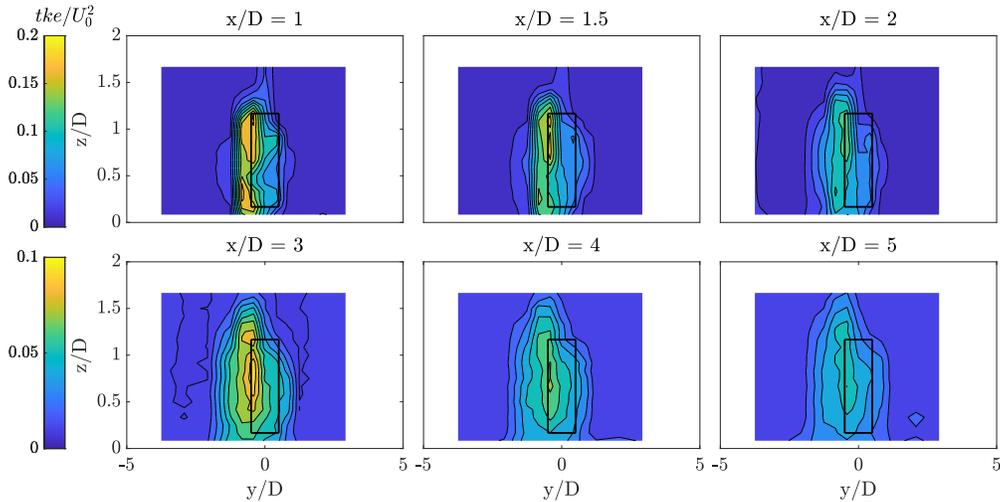


Figure 7: Contours of tke/U_0^2 at downstream cross sections located at $x/D = 1, 1.5, 2, 3, 4$ and 5 for the ST case, with the VAT rotating anti-clockwise. The solid black rectangle represents the perimeter of the turbine’s rotor. Flume’s centreline is at $y/D = 0$. Note that the range of values is adjusted for each row of contours corresponding at different streamwise locations.

233 the highest values of tke due to the turbine blades moving into the flow and energetic
 234 vortices being generated and shed [22]. Over the downstroke side, turbulence levels
 235 were significantly lower, likely linked to the reduced dynamic-stall vortices strength
 236 due to the Reynolds number of the experiments.

237 Turbulent momentum exchange is indicated by the horizontal and vertical com-
 238 ponents of the Reynolds shear stress ($\overline{u'v'}/U_0^2$, Figure 8, and $\overline{u'w'}/U_0^2$, Figure 9,
 239 respectively), which shows that regions of highest shear stresses were mostly found
 240 in the near wake. The high magnitudes observed for $\overline{u'v'}$, originated from the convec-
 241 tion of dynamic-stall vortices and interaction with the ambient flow, included both
 242 positive and negative values on the upstroke side ($y/D \leq 0$), with the latter found
 243 on the outside region of the turbine’s rotor swept area and the former $u'v'$ inside
 244 it. This pattern was observed only over the near-wake region as the turbine-induced
 245 vortical structures lose their coherence due to the mixing with the ambient flow.

246 Vertical Reynolds shear stresses ($\overline{u'w'}/U_0^2$) showed predominantly negative values
 247 on the upper half of the turbine ($z/D \geq 0.8$), due to the flow over-topping the turbine
 248 being transported downward into the wake. In the near wake ($x/D \leq 2$), positive
 249 $\overline{u'w'}$ values appeared on the lower half of the turbine ($z/D \leq 0.7$) with particularly
 250 large values near the corners of the rotor’s swept area, which unveils the interaction

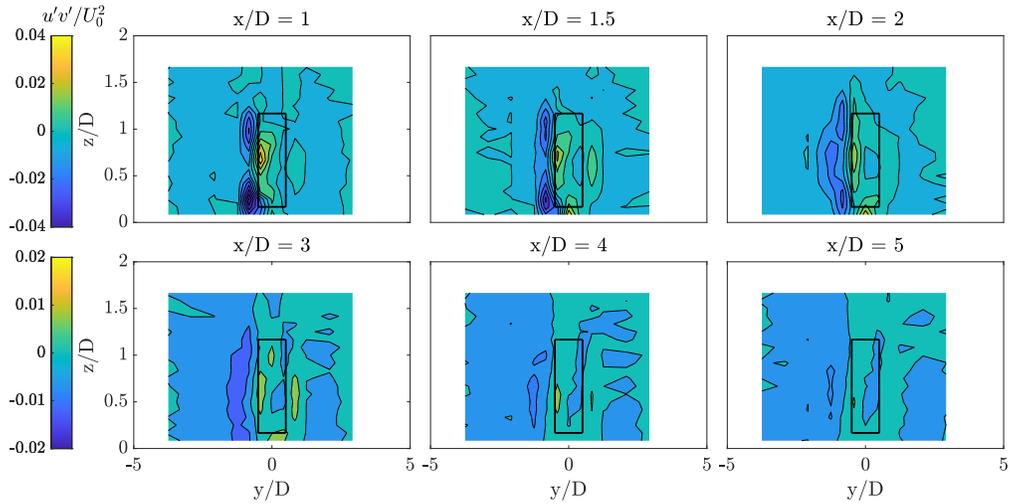


Figure 8: Contours of $\overline{u'v'}/U_0^2$ at downstream cross sections located at $x/D = 1, 1.5, 2, 3, 4$ and 5 for the ST case, with the VAT rotating anti-clockwise. The solid black rectangle represents the perimeter of the turbine's rotor. Flume's centreline is at $y/D = 0$. Note that the range of values is adjusted for each row of contours corresponding at different streamwise locations.

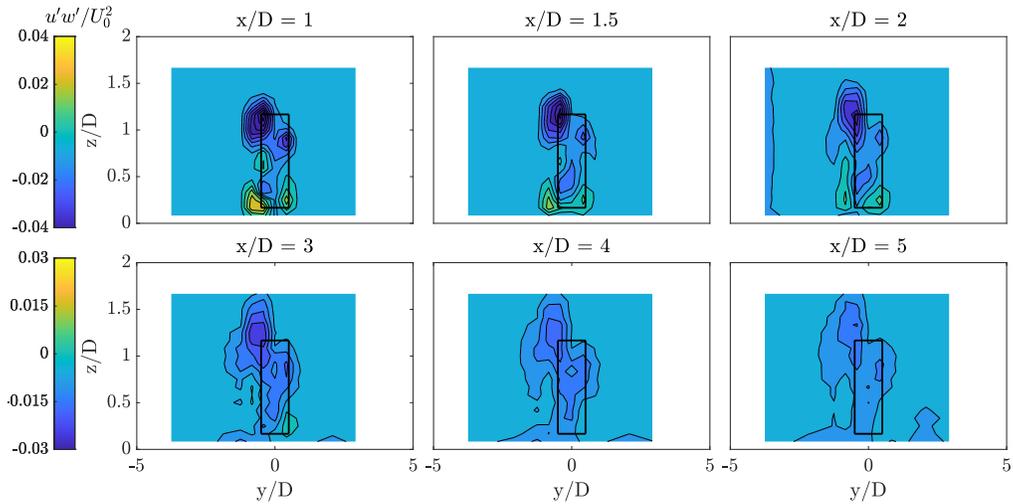


Figure 9: Cross section contours of $\overline{u'w'}/U_0^2$ at downstream locations of $x/D = 1, 1.5, 2, 3, 4$ and 5 for the ST case, with the VAT rotating anti-clockwise. The solid black rectangle represents the perimeter of the turbine's rotor. Flume's centreline is at $y/D = 0$. Note that the range of values is adjusted for each row of contours corresponding at different streamwise locations.

251 between the bottom tip vortices and the upward flow going through the bottom gap
 252 between the turbine’s rotor and flume bed. Further downstream after $x/D = 3$ again,
 253 $\overline{u'w'}$ shear stresses have significantly decayed due to the mixing of the wake with the
 254 ambient flow.

255 4.3. Twin-VAT wake results

256 The impact of rotational direction and lateral turbine spacing is now elucidated
 257 for each of the twin-VAT cases. The characteristics of the wakes developed down-
 258 stream of the six twin-VAT setups, i.e., for three rotation combinations and two inter-
 259 turbine spacings: SR–1.5, CRF–1.5, CRB–1.5, SR–2.0, CRF–2.0, and CRB–2.0, were
 260 analysed using contours at cross-sections normal to the flow direction ($y - z$ -planes)
 261 at several streamwise locations. Mean streamwise velocities are presented for cases
 262 adopting the two inter-turbine separations of $S_y/D = 1.5$ and 2.0 while higher-order
 263 statistics are discussed only for setups with $S_y/D = 1.5$, as those with $S_y/D =$
 264 2.0 have a similar spatial distribution and for brevity are included in the Appendix
 265 (Section 6).

266 Contours of \overline{u}/U_0 at $x/D = 1, 2, 3, 5$ and 10 are presented in Figure 10 for SR–
 267 1.5, CRF–1.5, and CRB–1.5, and in Figure 11 for SR–2.0, CRF–2.0, and CRB–2.0.
 268 Similar wake characteristics to the ST case (Figure 6) were found in individual wakes
 269 at $x/D = 1.0$ behind each of the turbines for the SR and CRF cases, with the lowest
 270 velocities found on the upstroke side of the blades rotation [17]. For CRF–1.5, the
 271 wakes appear asymmetric to the vertical axis between the turbines. However, in the
 272 CRB–1.5 case the wakes already merge at $x/D = 1.0$, as the low-momentum region
 273 of the individual wakes collapse.

274 In the SR–1.5 and SR–2.0 cases, the same rotational direction of both turbines
 275 caused the individual asymmetric wakes to progress alongside each other within the
 276 near-wake ($x/D \leq 2$), which is better depicted from the setup with $S_y/D = 2.0$.
 277 In the transition zone ($x/D \geq 2$) both wakes started to interact and merge into a
 278 single combined low-momentum region by $x/D = 5$, as represented in Figure 2 (a)
 279 and similar to [29]. In the far-wake, the velocities show that the combined wake
 280 expanded vertically across the water column and laterally, especially to the left-
 281 hand side of T_2 . In fact, the widest wake extent was found at $x/D = 10.0$ for this
 282 configuration.

283 The individual wakes in the CRF (counter-rotating forward) cases move outwards
 284 in opposite directions [17], with the bypass flow enhanced by the downstroke rotation
 285 of the blades which further separated both low-momentum wakes, especially for the
 286 CRF–2.0 setup, as illustrated in Figure 2 (b). Both wakes remained separated by
 287 the bypass flow until $x/D \approx 1.5$ and mirrored each other relative to the centreline

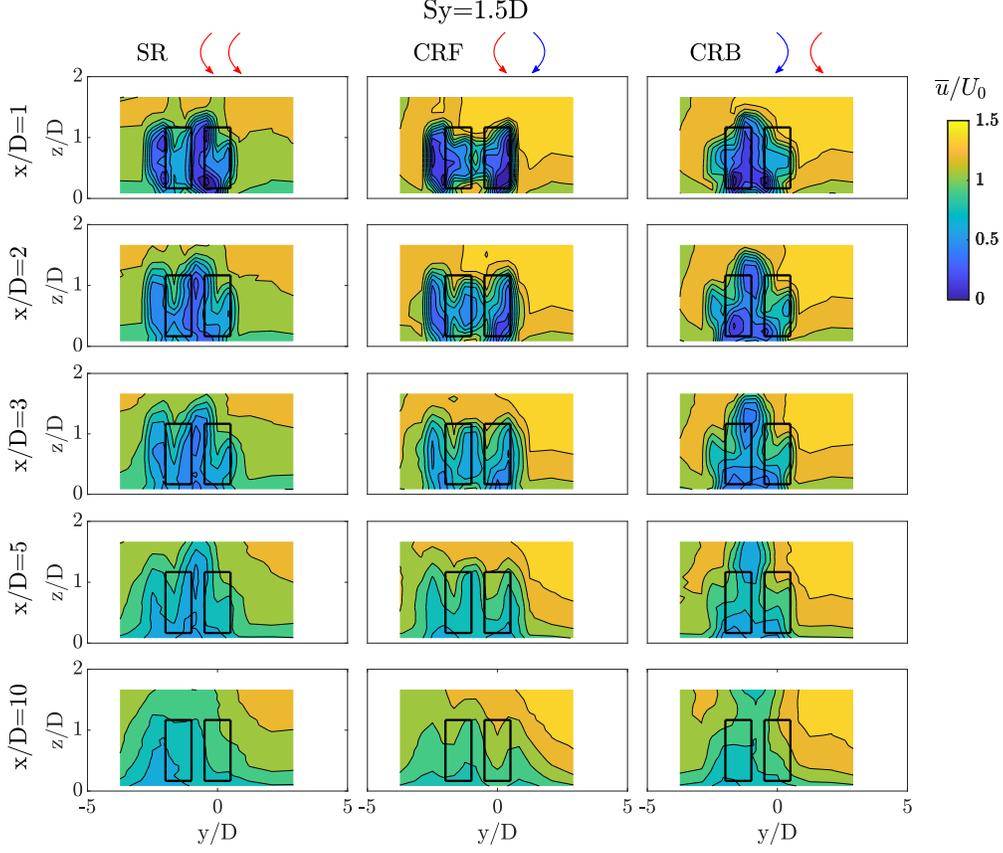


Figure 10: Cross section contours of \bar{u}/U_0 at downstream locations of $x/D = 1, 2, 3, 5$ and 10 for the SR-1.5 (left), CRF-1.5 (middle), and CRB-1.5 (right) cases. The solid black rectangles represent the perimeter of the turbine’s rotor and the flume’s centreline is located at $y/D = 0$.

288 through the gap spacing S_y . For CRF-1.5, the wakes gradually merged with further
 289 increasing downstream distance ($x/D \geq 5$), although for CRF-2.0 these appear to
 290 be more independent, likely due to the blades’ downstroke motion within the bypass
 291 region that further amplified the relative velocity of the flow through, i.e., $\bar{u}/U_0 \geq 1$,
 292 thus isolating both wakes and delaying their mixing. Similarly, [17] observed that
 293 both wakes were separated by the bypass flow until $x/D \approx 6$ for CRF twin-turbines
 294 spaced by $S_y/D = 1.2$, which is in agreement with the presented results for the
 295 CRF-1.5 case. For an inter-turbine spacing of $S_y/D = 2.0$, [29] reported almost
 296 no interaction between both wakes by $x/D = 10$ as observed in this study. The
 297 region of limited interaction between both turbine wakes suggests an ideal location

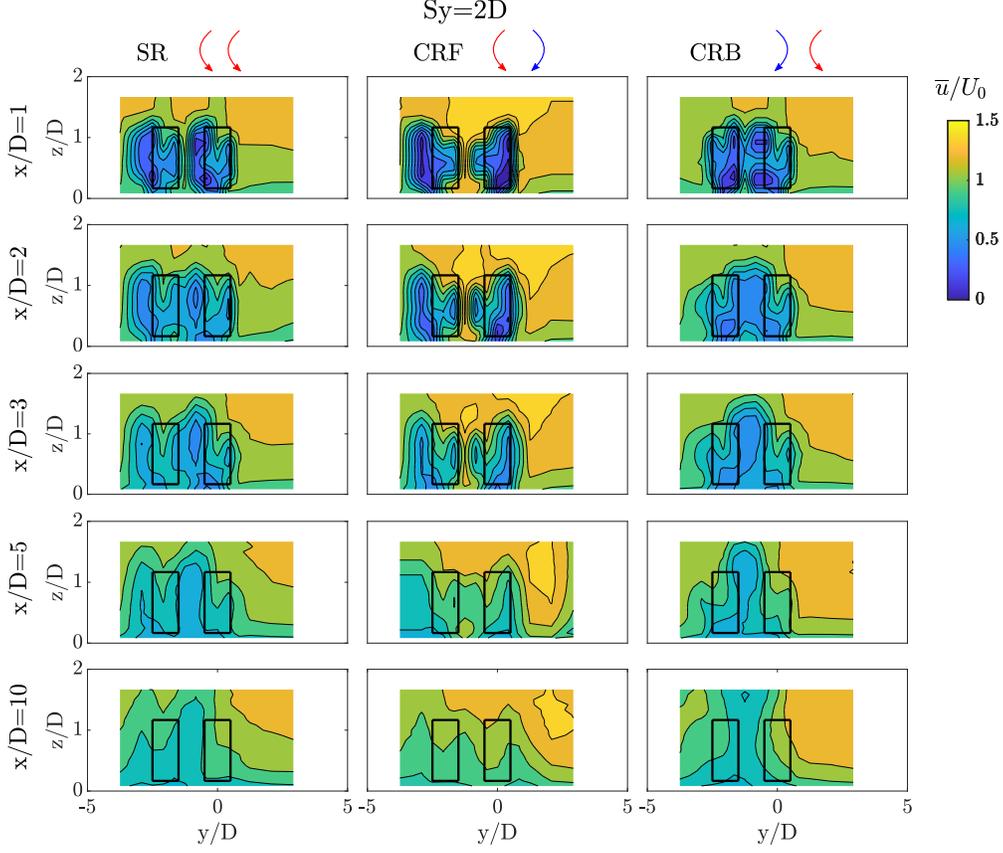


Figure 11: Cross section contours of \bar{u}/U_0 at downstream locations of $x/D = 1, 2, 3, 5$ and 10 for the SR-2.0 (left), CRF-2.0 (middle), and CRB-2.0 (right) cases. The solid black rectangles represent the perimeter of the turbine’s rotor and the flume’s centreline is located at $y/D = 0$.

298 for positioning a second row of turbines downstream.

299 Such limited interaction led to the wake in the CRF cases having the highest
 300 streamwise velocities at the furthest measured location of $x/D = 10$ compared to
 301 the other setups, as shown later in Section 4.4 in terms of spatially averaged velocity
 302 values.

303 In the CRB (counter-rotating backward) cases, individual turbine wakes inter-
 304 acted with one another shortly downstream of the rotors, collapsing into a single
 305 low-momentum region at $x/D = 1.0$ for the CRB-1.5 case whilst two wakes were
 306 observed for CRB-2.0 at $x/D = 1.0$ which merged at $x/D = 2.0$. Similarly, in [29]
 307 wakes began to merge at $x/D = 3.0$ and completely merged by $x/D = 7.0$ for a sim-

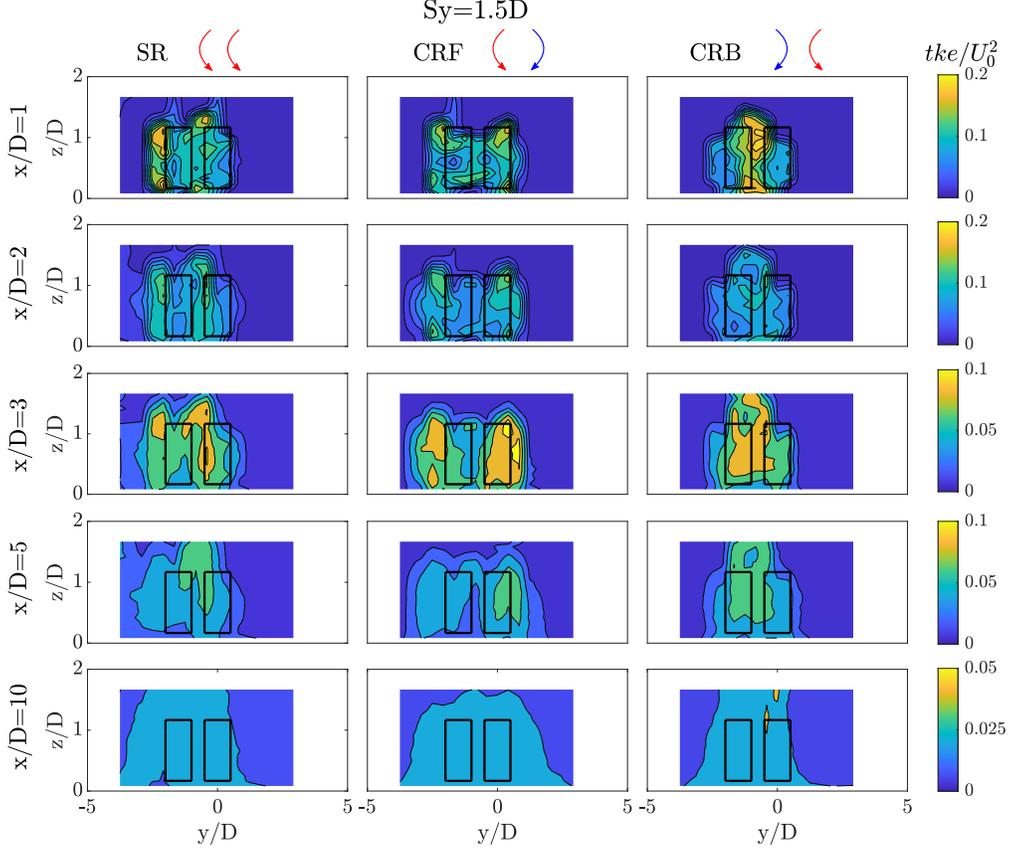


Figure 12: Cross section contours of tke/U_0^2 at downstream locations of $x/D = 1, 2, 3, 5$ and 10 for the SR-1.5 (left), CRF-1.5 (middle), and CRB-1.5 (right) cases. The solid black rectangles represent the perimeter of the turbine’s rotor. The flume’s centreline is located at $y/D = 0$. Legend scale is adjusted at every streamwise location (x/D) to ease depiction of hydrodynamic features.

308 ilar CRB-2.0 case. Such collapse into a single wake resulted from a weaker bypass
 309 flow as in this region the blades moved into the flow (upstroke rotation) [29]. After
 310 $x/D = 3.0$, the combined wake occupied a narrower lateral extent compared to SR
 311 and CRF cases, extended notably in the vertical direction throughout most of the
 312 water column, and was nearly axisymmetric relative to the vertical axis at the centre
 313 of the combined swept area, i.e., $S_y/2$.

314 The distribution of tke/U_0^2 (Figure 12) for each of the $S_y/D = 1.5$ configurations
 315 was found to be similar to that for the ST case (Figure 7), i.e. the areas with highest
 316 tke pockets are those with the lowest velocity magnitude. During the upstroke
 317 movement, the blades shed vortical structures that increased turbulent mixing and

318 triggered high levels of tke/U_0^2 . Although both wakes in the SR-1.5 case evolve
 319 independently in the near-wake region, their interaction and merging in the transition
 320 zone ($x/D \geq 2$) resulted in a region of high turbulent kinetic energy behind the twin-
 321 turbine swept area, which enhanced the mixing of both wakes [29]. Particularly high
 322 values of tke/U_0^2 were observed in the bypass region immediately downstream of the
 323 CRB-1.5 case due to the collapse of the wake regions generated from the upstroke
 324 motion of the blades. In this region, the tke values for CRF-1.5 were reduced, with
 325 maxima located on the outskirts of the wake. With increasing downstream distance,
 326 the turbulent kinetic energy decreased to approach the values of the upstream tke
 327 levels at $x/D = 10$.

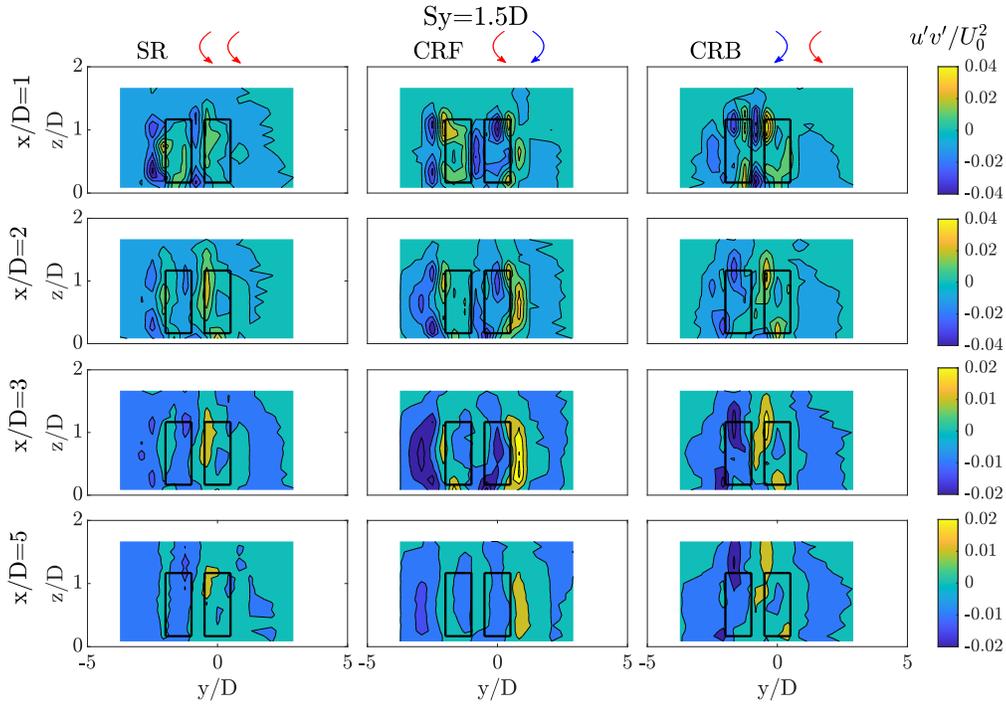


Figure 13: Cross section contours of $\overline{u'v'}/U_0^2$ at downstream locations of $x/D = 1, 2, 3,$ and 5 for the co-rotating (SR-1.5; left), counter-rotating forward (CRF-1.5; middle), and counter-rotating backward (CRB-1.5; right) with $S_y/D = 1.5$. The solid black rectangles represent the perimeter of the turbine's rotor. The flume's centreline is located at $y/D = 0$. Legend scale is adjusted for different streamwise locations (x/D) to ease depiction of hydrodynamic features.

328 Reynolds shear stresses $\overline{u'v'}/U_0^2$ and $\overline{u'w'}/U_0^2$ are presented in Figures 13 and 14,
 329 respectively, for all $S_y/D = 1.5$ twin-VAT cases. Peak $\overline{u'v'}/U_0^2$ magnitudes were ob-

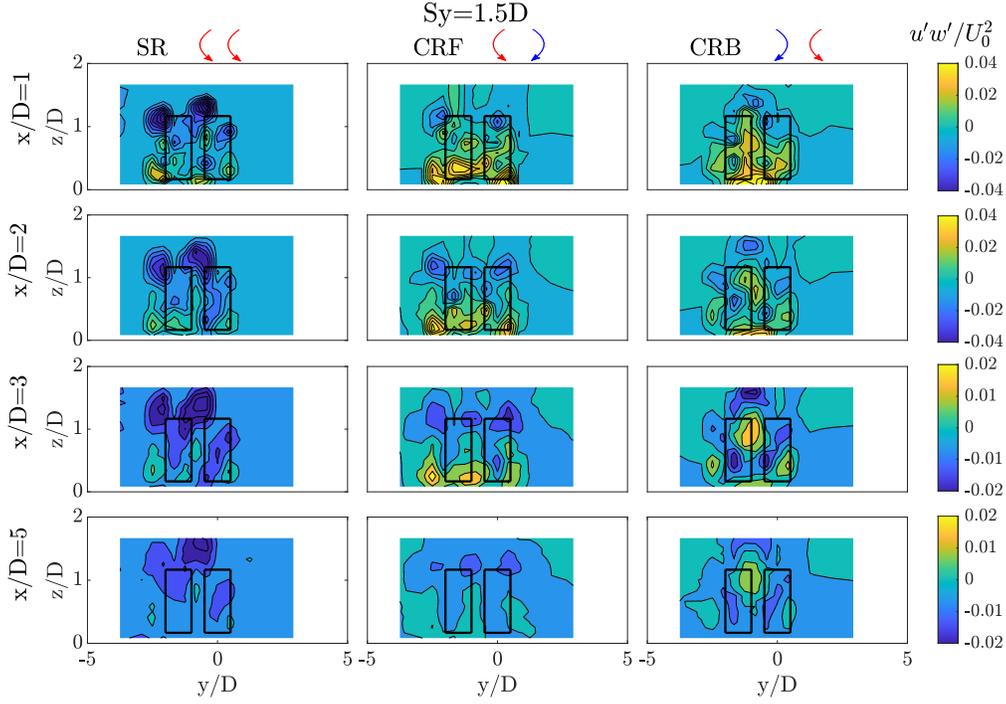


Figure 14: Cross section contours of $\overline{u'w'}/U_0^2$ at downstream locations of $x/D = 1, 2, 3,$ and 5 for the SR-1.5 (left), CRF-1.5 (middle), and CRB-1.5 (right) cases. The solid black rectangles represent the perimeter of the turbine's rotor. The flume's centreline is located at $y/D = 0$. Legend scale is adjusted for different streamwise locations (x/D) to ease depiction of hydrodynamic features.

330 served on the periphery of the swept areas of each turbine, indicative of turbulent
 331 momentum exchange and where vortices are generated. These are clearly observed
 332 for the CRF-1.5 case even up to $x/D = 5.0$. Similar to the ST case, vertical Reynolds
 333 shear stresses $\overline{u'w'}/U_0^2$ were mostly negative on the upper half of the turbines as tur-
 334 bulent momentum entrained downwards into the wake region, and positive on the
 335 lower half; with the exception of CRB-1.5 in which the mixing of vortical struc-
 336 tures from each of the turbine rotors showed a different pattern suggesting that the
 337 turbulent wake flow is more complex for this setup. Tip vortices were also present,
 338 triggering high shear stress levels around the top tips of the blades in the near wake
 339 ($x/D \leq 2$).

340 4.4. Wake recovery

341 The integral change of the wake in the downstream direction was estimated in
 342 terms of the cross-sectional average of the streamwise velocity and turbulence inten-

343 sity. These were approximated by integrating the measured quantities at the ADV
 344 locations within the turbine area for ST case ($0.5 \leq y/D \leq 0.5$) and the region
 345 spanning both VATs for the twin setups ($S_y/D = 1.5$: $0.5 \leq y/D \leq 2$, $S_y/D = 2.0$:
 346 $0.5 \leq y/D \leq 2.5$). In the vertical, only those points within the turbine area were
 347 considered. The spatial-averaging operation is denoted as $\langle \cdot \rangle$ and results of $(\langle \bar{u} \rangle / U_0)$
 348 and $(\langle u' \rangle / U_0)$ are provided in Figures 15 and 16 for the ST and six twin-VAT cases
 349 comparing rotational direction and lateral spacing, respectively.

350 Immediately downstream of the turbines ($x/D = 1$), the cross-sectional mean
 351 velocity recovery was observed to exceed values of $\langle \bar{u} \rangle / U_0 \geq 50\%$ for all configura-
 352 tions, especially for CRF (counter-rotating forward) configurations which attained
 353 the highest initial wake velocity. Larger intra-turbine spacing consistently enhanced
 354 wake recovery due to a higher momentum flowing through the bypass region, e.g.,
 355 in the case of CRF-2.0, $\langle \bar{u} \rangle / U_0$ fully recovered the bulk velocity value at $x/D = 5$,
 356 while for CRF-1.5 this was at approximately $x/D = 8$. Vergaerde et al. [17] adopted
 357 an inter-turbine spacing of $S_y/D = 1.3$ reporting a wake recovery of 75% at $x/D =$
 358 5.2.

359 Figure 16 indicates that rotational direction plays a more important role than
 360 lateral spacing shown in Figure 15 for the current values of S_y . In comparison to
 361 the single turbine case, CRF configurations featured the largest kinetic energy in the
 362 wake region due to a higher initial wake velocity, even exceeding the velocities from
 363 the single turbine wake. CRB setups followed a similar wake velocity evolution over
 364 the wake length as the single turbine but with full wake recovery attained at 8 and
 365 10 diameters downstream, respectively. The slowest wake recovery was found for SR
 366 cases that achieved velocities of $\langle \bar{u} \rangle / U_0 \geq 80\%$ at $x/D = 10$ despite featuring larger
 367 velocities than the single turbine case at $x/D = 1$.

368 In terms of wake unsteadiness, the highest turbulence intensity values $(\langle u' \rangle / U_0)$
 369 were found for both CRB (counter-rotating backward) cases as a consequence of
 370 the large interaction between both wakes at all downstream locations, while the
 371 lowest turbulence intensities were found in the case of CRF-2.0 due to the wider
 372 inter-turbine spacing that minimised the interplay between turbulent wakes. Some
 373 variability in the spatially-averaged turbulence intensity is observed at $x/D = 1$, 1.5
 374 and 2 for configurations with $S_y/D=2.0$, agreeing with the contours shown in Figure
 375 20. In all cases, free-stream values of $\langle u' \rangle / U_0$ were reached at approximately $x/D =$
 376 10.

377 The rate of wake recovery of the seven configurations is presented in Figure 17,
 378 showing the spatially-averaged velocity deficit ($\Delta \langle \bar{u} \rangle$) and decay slopes of -1/3, -1/2
 379 and -2/3. Classic shear-flow theory states that for self-similar axisymmetric and
 380 planar wakes the velocity deficit decay should be proportional to $x^{-2/3}$ and $x^{-1/2}$,

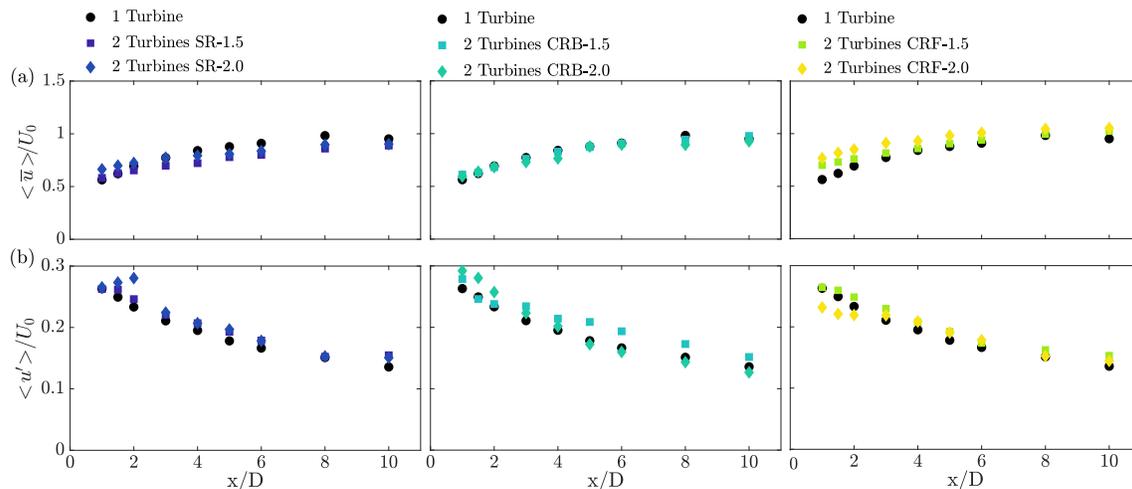


Figure 15: Comparison of the lateral spacing impact for the three rotational directions with values of spatially-averaged (a) mean streamwise velocity $\langle \bar{u} \rangle$, and (b) turbulent intensity $\langle u' \rangle$, normalised by U_0 at all measured locations in downstream direction.

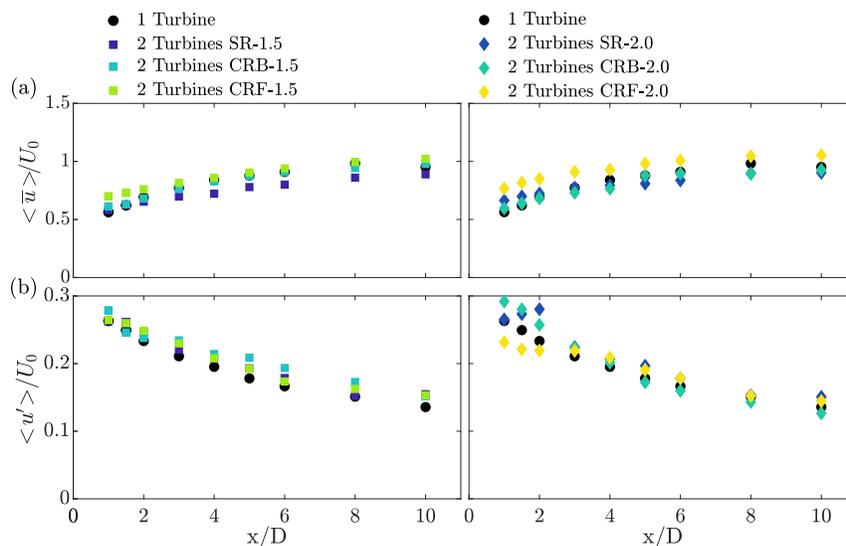


Figure 16: Comparison of the rotational direction for the two lateral spacing values with values of spatially-averaged (a) mean streamwise velocity $\langle \bar{u} \rangle$, and (b) turbulent intensity $\langle u' \rangle$, normalised by U_0 at all measured locations in downstream direction.

381 respectively [35]. Here, whilst VAT wakes did not attain self-similarity within the
 382 measured range of $1 \leq x/D \leq 10$, the decay rates were between $-1/3$ and $-2/3$ for
 383 all cases. The single turbine (ST) and both CRB cases followed a $-1/2$ slope over
 384 the measured wake length. In contrast, the SR cases showed an initial slope approx.
 385 equal to $-1/3$ until $x/D = 4-5$ downstream when the slope increases and is closer
 386 to a $-1/2$ decay. CRF setups featured the slowest decay rates of $-1/3$ over the wake
 387 length, although these configurations showed the lowest velocity deficit at the wake
 388 onset.

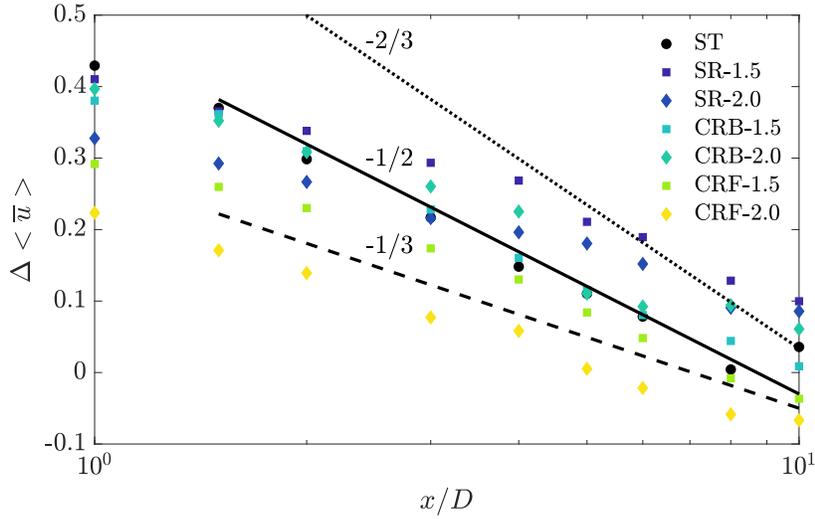


Figure 17: Wake recovery rate obtained from the spatially-averaged velocity deficit $\Delta \langle \bar{u} \rangle$ in semi-log scale for the seven cases. Straight, dashed and dotted lines represent the $-1/2$, $-1/3$ and $-2/3$ slopes.

389 Overall the results presented suggest that a lateral spacing of, at least, $2D$ should
 390 be considered with a counter-rotating forwards (CRF) configuration to enhance wake
 391 recovery when designing arrays of VATs with a minimum of two turbines per row
 392 to minimise detrimental wake effects. Further research, however, will be required
 393 to identify the optimal lateral turbine spacing. Moreover, configurations with a lat-
 394 eral spacing of $2D$ attained a faster wake recovery and experienced lower turbulence
 395 intensity. This suggests that in arrays with greater inter-turbine spacing, the pro-
 396 duction of turbulent kinetic energy is enhanced as a result of the destruction of the
 397 kinetic energy of the mean flow that delays the momentum recovery rate. As per
 398 the rotational direction of the twin-turbines, counter-rotating forwards leads to the
 399 largest velocities in the wake region and thus there is a larger kinetic energy to be

400 extracted by secondary rows.

401 4.5. Wake superposition

402 To provide further insights into the interaction between individual wakes in the
403 twin configurations, the streamwise velocity deficit ($\Delta\bar{u} = 1 - \bar{u}/U_0$) of the wake
404 from the ST case is superimposed using linear ($\Delta\bar{u}_{lin} = \Delta\bar{u}_{T_1} + \Delta\bar{u}_{T_2}$) and quadratic
405 ($\Delta\bar{u}_{quad} = [(\Delta\bar{u}_{T_1})^2 + (\Delta\bar{u}_{T_2})^2]^{0.5}$) superposition and compared to the actual mea-
406 sured values obtained for the six twin configurations.

407 Velocity deficit results at turbine mid-height ($z/D = 0.67$) for five downstream
408 locations for the SR-1.5 and SR-2.0 are shown in Figure 18, together with the linear
409 and quadratic wake superposition predictions, and values of the ST case. Within
410 the near wake region ($x/D \leq 2$), both wakes have not yet interacted with each
411 other, resulting in a good agreement from their individual superposition. Further
412 downstream from the turbines ($x/D \geq 4$), the velocity deficit in the wake of T_2 for
413 SR-1.5 was slightly underestimated whilst at $y/D \approx 1.0$ the linear model provides
414 a better fit than the quadratic superposition which overpredicts $\Delta\bar{u}$. The observed
415 small deviation between the superimposed wakes and actual measurements indicates
416 that cumulative flow effects on the evolution and merging of interacting twin-VAT
417 wakes are well represented with simple superposition techniques at all locations.

418 The root-mean-square (rms) error of the spatially-averaged mean streamwise ve-
419 locity ($\langle\bar{u}\rangle$) between the linear and quadratic superposition predictions and measured
420 experimental results were computed at each streamwise location for the six twin-VAT
421 configurations (Figure 19). Independent of the lateral spacing, a lower rms error was
422 found for SR cases, indicating that the developed combined wakes have reduced non-
423 linear interactions, allowing the superposition models to provide good predictions of
424 the velocity field. However, for CRB (counter-rotating backward) setups, the super-
425 position methods show a larger sensitivity to the lateral spacing (S_y/D) with rms
426 values for CRB-1.5 being higher than those for CRB-2.0 over the entire wake length.
427 In the latter case, the turbine wakes are closer and thus there is a higher degree of
428 interaction between them.

429 Considering the CRF cases, the accuracy of the superposition models remains
430 almost unchanged for the two intra-turbine spacing values. In comparison, rms
431 values of spatially averaged velocity reduced for $S_y/D = 2.0$ due to a wider spacing
432 between devices and thus a limited wake interaction (Figure 11). Overall, the linear
433 approach provides lower rms errors than the quadratic superposition when predicting
434 the mean velocity field.

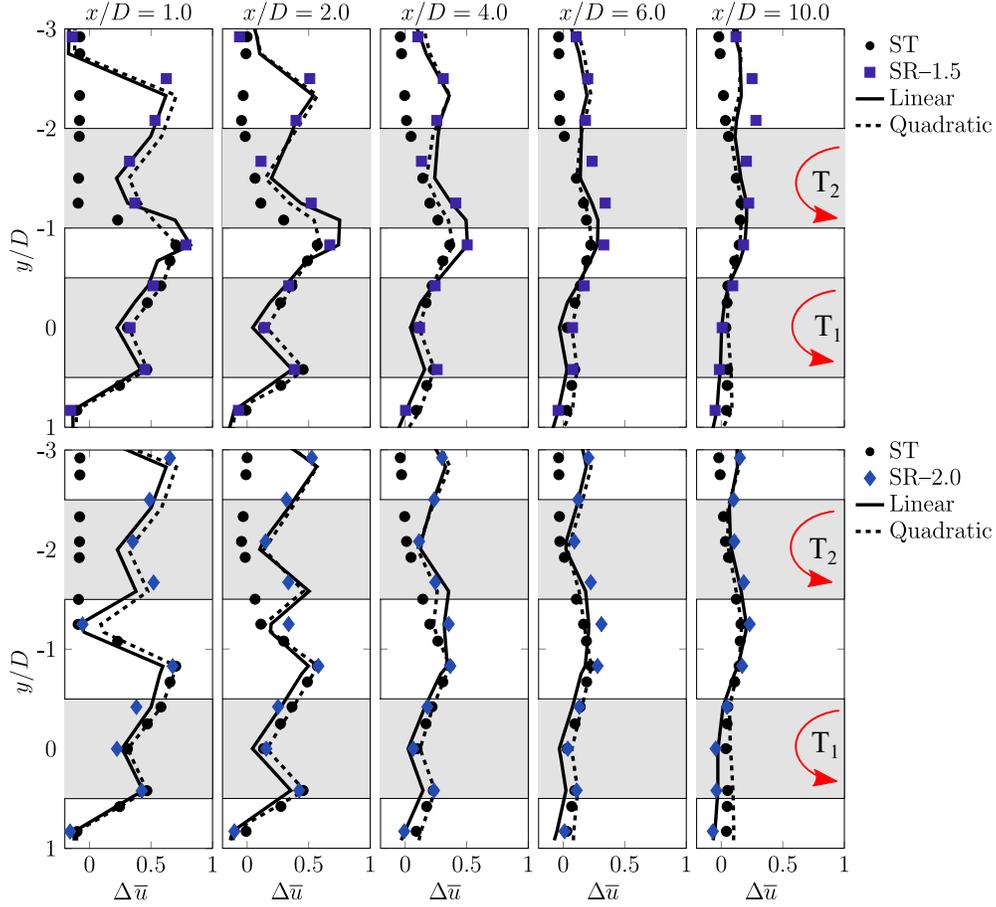


Figure 18: Results of velocity deficit ($\Delta\bar{u}$) of the SR-1.5 (top) SR-2.0 (bottom) cases with the calculation from the linear and quadratic superposition models. ST case data is included for convenience. Shaded areas denote the projected turbine's rotor swept area. The centreline of the flume is located at $y/D=0$.

435 5. Conclusion

436 The evolution and interaction between the wakes of adjacent Vertical Axis Tur-
 437 bines (VATs) deployed in twin configurations has been experimentally studied by
 438 means of acoustic Doppler velocimeter (ADV) measurements. A standalone and six
 439 twin-VAT setups were tested, including shaft-to-shaft spacings (S_y) of 1.5 and 2.0
 440 turbine diameters (D) with the devices rotating in the same and in opposite direc-
 441 tions. The results presented show that the wake evolution was more sensitive to
 442 the rotational direction of the VATs than their lateral spacing for the range of S_y

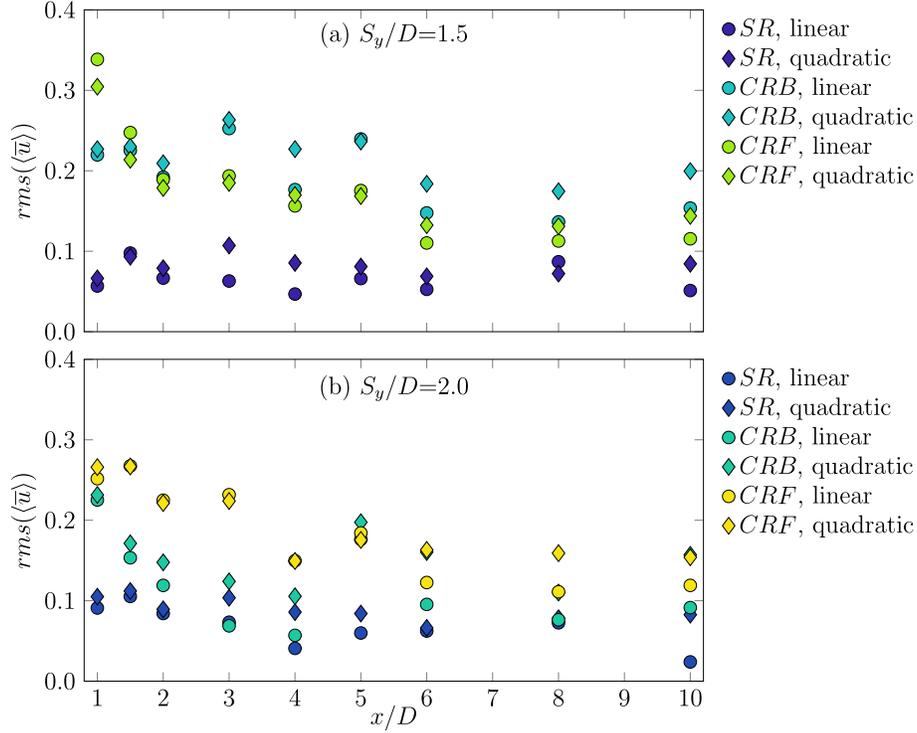


Figure 19: Root-mean-square (rms) error in the prediction of the spatially averaged mean streamwise velocity $\langle \bar{u} \rangle$ adopting linear and quadratic superposition methods for the twin-turbine cases with $S_y/D = 1.5$ (a) and 2.0 (b).

443 values adopted. When VATs rotated in the same direction (SR cases), the wake re-
 444 gion notably expanded in the lateral and vertical directions. In the counter-rotating
 445 forwards setups (CRF, i.e., turbine blades move along with the flow in the bypass
 446 region), the highest momentum deficit was found in the downstroke regions on the
 447 outskirts of the wake, preventing the wakes from merging and therefore, allowing
 448 larger velocities in the wake. In the counter-rotating backwards cases (CRB, i.e.,
 449 blades move against the flow in the bypass region), the low-velocity regions merged
 450 immediately downstream of the turbines, generating a single wake that was relatively
 451 narrow in the lateral direction but expanded over the whole water column after 3
 452 or $5D$ downstream of the turbines depending on whether the intra-turbine spacings
 453 was $1.5D$ or $2.0D$, respectively. Similar distribution patterns were observed in the
 454 turbulent kinetic energy and Reynolds shear stresses.

455 Cross-sectional averaged values of mean velocity and turbulence intensity outlined

456 that cases with turbines rotating in a counter-rotating forward (CRF) sense achieved
457 the highest momentum in the wake. This was specially noticeable for the CRF-2.0
458 configuration in which the cross-sectional mean velocity reached the bulk velocity
459 value (U_0) at $5D$ downstream, a distance noticeably shorter than the $8D$ at which
460 the single turbine attained full wake recovery. The rate of momentum deficit decay
461 was slowest for the latter CRF cases, being proportional to $(x/D)^{-1/3}$, whilst for
462 the CRB cases and the single turbine featured similar wake velocities with a faster
463 recovery rate close to $(x/D)^{-1/2}$. Conversely, SR setups achieved the lowest velocities
464 in the wake, with a value of $90\%U_0$ at the furthest measured location of $x/D = 10$,
465 with the velocity deficit recovering at an approx. rate of $(x/D)^{-1/3}$ until $4-5D$ after
466 which the decay accelerated with a $-1/2$ slope.

467 The momentum recovery was enhanced and turbulence intensity decreased with
468 increasing intra-turbine spacings, i.e., adopting $S_y/D = 2$, suggesting that greater
469 turbine spacings may be beneficial when designing multi-row twin-VAT arrays. Fur-
470 ther research, however, will be required to identify the optimal lateral distance be-
471 tween turbines that maximises installed power density per unit land, i.e., taking into
472 account reduced wake effects to enhance the performance of downstream turbines
473 while allowing the twin-turbines to increase device energy yield due to synergistic
474 blockage effects.

475 Despite the complexity of the wake dynamics, adopting a linear or quadratic
476 superposition of a single turbine wake in the horizontal plane appeared to yield good
477 results when compared to experimentally measured mean velocity values. In terms of
478 root-mean-square errors, these were higher for turbines rotating in a counter-rotating
479 manner as the dynamics of the interacting wakes might result in more non-linear
480 effects.

481 This study provides new insights into the wake characteristics behind twin-VAT
482 arrays and informs the design of future multi-row arrays of VATs with minimised
483 wake-turbine interactions. The measured wakes suggest that a lateral spacing of
484 $2D$ with a counter-rotating forward (CRF) setup would allow to adopt a streamwise
485 spacing between rows of $5D$ so that secondary rows can harness kinetic energy ef-
486 ficiently. On the other hand, adopting configurations with turbines rotating in the
487 counter-rotating backwards (CRB) and same rotation (SR) sense would require a
488 wider streamwise inter-row spacing of, at least, $10D$ unless the lateral turbine spac-
489 ing is increased which, however, would decrease the installed power density capacity.

490 **Acknowledgement**

491 This research was financially supported using seed funding from Cardiff Univer-
492 sity's GCRF QR Funding from the Higher Education Funding Council for Wales,

493 under the project "HEFCW GCRF Small Project SP111: A new technology of fish-
494 friendly river turbines to provide renewable energy to impoverished isolated commu-
495 nities". In addition, SM was funded as part of the Water Informatics Science and
496 Engineering Centre for Doctoral Training (WISE CDT) under grant EP/L016214/1
497 from the Engineering and Physical Science Research Council (EPSRC). The authors
498 are grateful to Dr Aldo Benavides, Dr Carlos Duque and Maxime Lacennec for the
499 fruitful conversations about this research, and Paul Leach for his invaluable support
500 to build the turbine prototypes and technical assistance during the measurements.
501 Data underpinning the results presented here can be found in the Cardiff University
502 data catalogue: <http://doi.org/10.17035/d.2021.0134567672>. Comments from
503 the Editor and three anonymous reviewers are greatly appreciated.

504 **6. Appendix**

505 Here, the results obtained for the twin-VAT configurations with $S_y/D = 2.0$ are
506 presented in terms of turbulent kinetic energy (Figure 20), horizontal (Figure 21)
507 and vertical (Figure 22) Reynolds shear stresses. In these figures, the rectangular
508 outline indicates the swept area of the turbines.

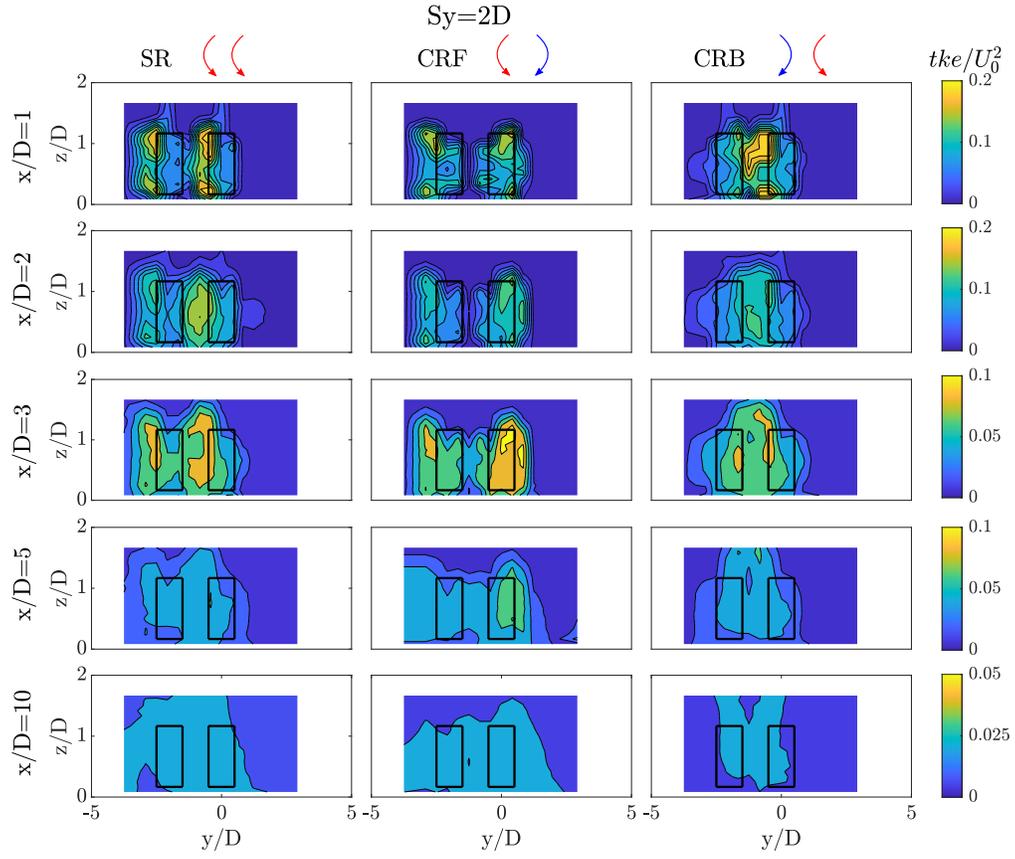


Figure 20: Cross section contours of tke/U_0^2 at downstream locations of $x/D = 1, 2, 3,$ and 5 for the SR-2.0, CRF-2.0 and CRB-2.0. The solid black rectangles represent the perimeter of the turbine's rotor. The flume's centreline is located at $y/D = 0$. Legend scale is adjusted for different streamwise locations (x/D) to ease depiction of hydrodynamic features.

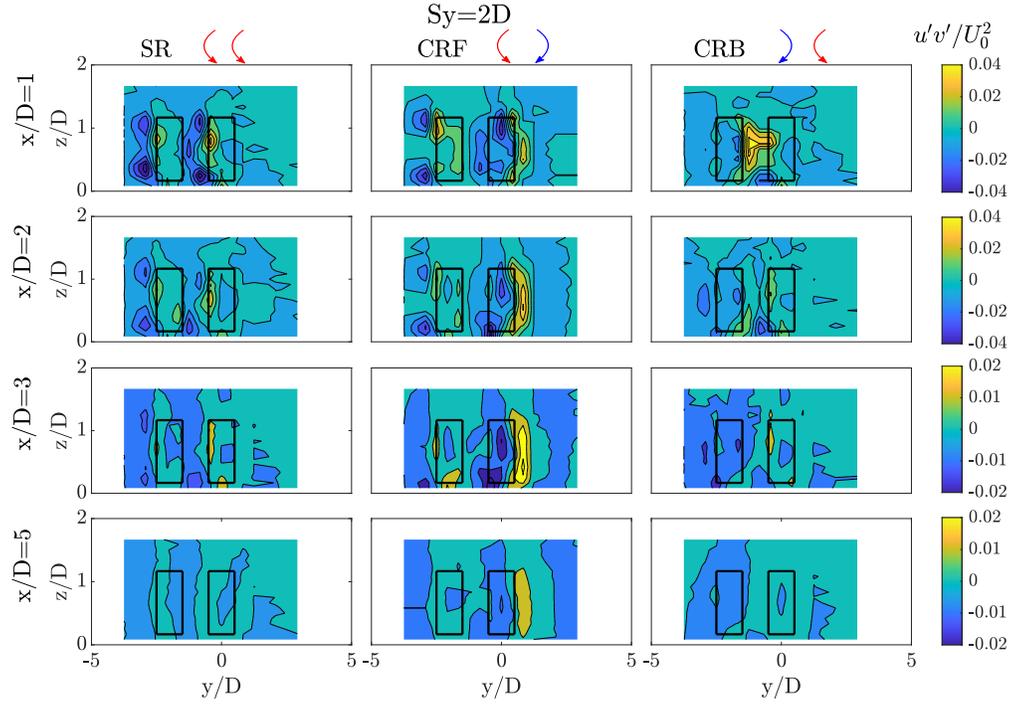


Figure 21: Cross section contours of $\overline{u'v'}/U_0^2$ at downstream locations of $x/D = 1, 2, 3,$ and 5 for the SR-2.0, CRF-2.0 and CRB-2.0. The solid black rectangles represent the perimeter of the turbine's rotor. The flume's centreline is located at $y/D = 0$. Legend scale is adjusted for different streamwise locations (x/D) to ease depiction of hydrodynamic features.

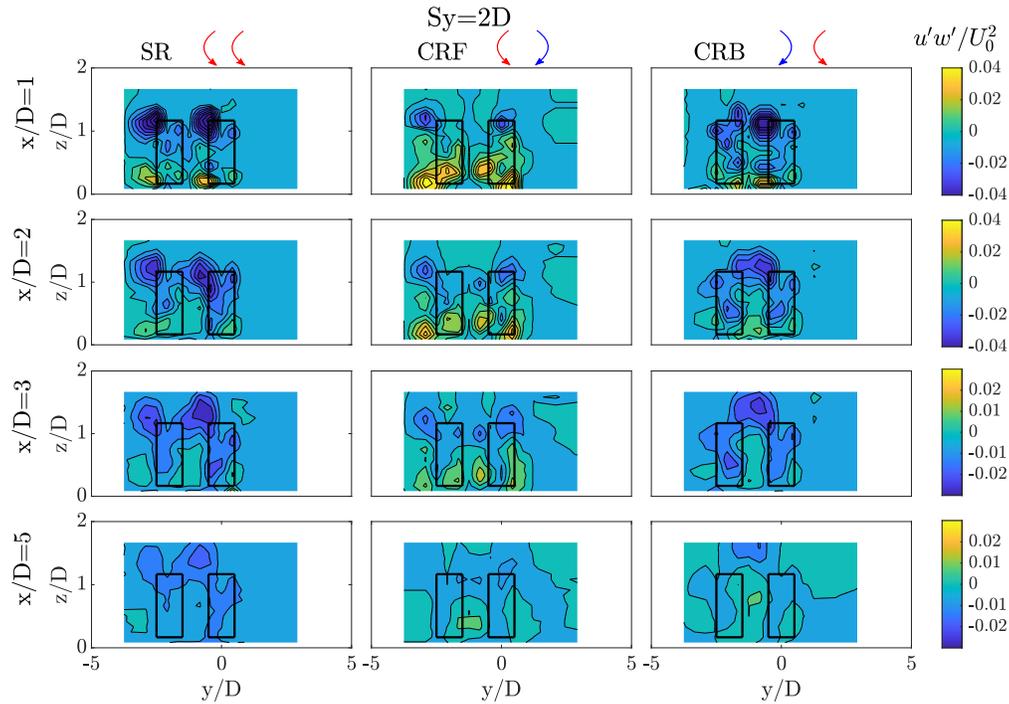


Figure 22: Cross section contours of $\overline{u'w'}/U_0^2$ at downstream locations of $x/D = 1, 2, 3,$ and 5 for the SR-2.0, CRF-2.0 and CRB-2.0. The solid black rectangles represent the perimeter of the turbine’s rotor. The flume’s centreline is located at $y/D = 0$. Legend scale is adjusted for different streamwise locations (x/D) to ease depiction of hydrodynamic features.

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