



## Research article

# Nonlinear robust sliding mode - Backstepping hybrid control for WECS -theoretical design and experimental evaluation

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## ABSTRACT

This paper proposes a new contribution in the field of optimizing control techniques for wind systems to enhance the quality of the energy produced in the grid. Although the Sliding Mode control technique, whether classical or involving the use of artificial intelligence, has shown interesting results, its main drawback lies in the oscillation phenomenon commonly referred to as “chattering.” This phenomenon affects the accuracy and robustness of the system, as well as the parametric variation of the system. In this work, we propose a solution that combines two nonlinear techniques based on the Lyapunov theorem to eliminate the chattering phenomenon. It is a hybrid approach between the Backstepping strategy and the Sliding Mode, aiming to control the active and reactive powers of the doubly fed induction generator (DFIG) connected to the electrical grid by two converters (grid side and machine side). This hybrid technique aims to improve the performance of the wind system in terms of precision errors, stability, as well as active and reactive power. The proposed solution has been validated in the Matlab & Simulink environment to assess the performance and robustness of the proposed model, as well as experimentally validated on a test bench using the DSPACE 1104 card. The obtained results are then compared with other techniques, demonstrating a significant improvement in performance.

$V$	Wind speed (m/s)	$C_{t-est}$	Estimated torque of the turbine
$P_{aero}$	Aerodynamic power	$M$	Cyclic mutual inductance between stator and rotor
$P_{mec}$	Mechanical power	$C_{em}$	Electromagnetic torque
$R_g$	Radius of the turbine, Gearbox gain	$C_{aero}$	Aerodynamic torque
$f_s, f_r$	Stator and Rotor frequency	$f$	Coefficient of viscous friction
$P_s, P_r$	Stator and Rotor active power	$J_{tot}, J_t, J_g$	Total inertia, Turbine inertia and inertia of generator
$Q_s, Q_r$	Stator and Rotor reactive power	$\psi_{s,rd}, \psi_{s,rq}$	Stator and Rotor Flux in plane d – q d-q
$P_c$	Active power at the capacitor	$V_{s,rd}, V_{s,rq}$	Stator and Rotor Voltage in plane d – q

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(continued)

$\Omega_s, \Omega_r$	Synchronism speed and Rotor speed	$V_{fd}, V_{fq}$	Filter Voltage in plane d – q
$\Omega_t$	Turbine speed	$I_{sd}, I_{sq}$	Stator Current in plane d – q
$\omega_s, \omega_r$	Pulsation at stator and rotor	$I_{rd}, I_{rq}$	Rotor Current in plane d – q
$g$	Sliding of asynchronous machine	$I_{fd}, I_{fq}$	Filter Current in plane d – q
$p$	Number of poles pairs	$V_{dc}$	DC-BUS voltage
$R_s, R_r$	Stator and Rotor resistance	$S_{s,abc}$	Switching states of inverter
$L_s, L_r$	Stator and Rotor inductance	$P_f, P_r$	Filter and rotor active power
$R_f, L_f$	Filter resistance, Filter inductance		
$\omega_g$	Grid pulsation		

## 1. Introduction

### 1.1. Overview

In the ever-evolving story of global energy, the winds of change are pushing us towards a cleaner, more sustainable future [1]. As the world grapples with the urgent need to address climate change and reduce dependence on finite fossil fuels [2], renewable energy sources [3,4] have emerged as beacons of hope. Among these, wind power stands out, harnessing the kinetic force of the wind to generate electricity and redefining the energy landscape [5].

The temperature and pressure differences caused by the absorption of solar rays in the atmosphere ensure that wind energy is an indirect form of solar energy. The movement of air masses accumulates kinetic energy, which can be transformed and used for a number of purposes, including the production of electrical energy. Three factors affect how much electricity a wind turbine can generate: the size and design of the blades, the wind speed, and the air density due to temperature. The amount of energy that can be recovered by a wind turbine depends on the area that its rotor sweeps and the cube of the wind speed [6].

Wind power, a form of renewable energy with ancient roots in sail-driven ships, has undergone a remarkable metamorphosis in recent decades. Today, it stands as a symbol of innovation and progress, contributing significantly to the diversification of our energy mix. Wind turbines, resembling modern-day giants with their towering blades, have become iconic symbols of sustainable energy production [7].

In the face of climate change and the imperative to reduce carbon emissions, countries worldwide are embracing wind power as a vital component of their energy portfolios [8]. International collaborations and agreements underscore the global commitment to transition towards sustainable practices, with wind power emerging as a key player in this collective effort [9]. Recent studies [10–12] have tried to prove that wind power is growing the fastest in the world in terms of development due to recent new renewable energy technology.

According to the European Parliament's Fact Sheets on the European Union [13], EU legislation on the promotion of renewable energies has evolved considerably over the last fifteen years. In 2018, EU leaders set a target of 32 % renewable energy in final energy consumption by 2030. In March 2023, in line with the EU's ambition to achieve climate neutrality by 2050, the co-legislators decided to increase the share of renewable energy to be achieved by 2030 to 42.5 %, with the hope of reaching 45 %. The updated Renewable Energy Policy Framework for 2030 and beyond is currently being negotiated.

In the present day, the realm of wind power is experiencing a surge in both technological advancements and global adoption [14]. The efficiency and capacity of wind turbines have soared to unprecedented heights, breaking barriers and setting new standards [15]. This surge is not confined to a single region; it's a global phenomenon that transcends borders. From wind-swept plains to coastal shores, wind farms are sprouting up, painting a hopeful picture of a planet transitioning towards cleaner, greener energy solutions.

Beyond environmental benefits [16], wind power is proving its economic mettle [17]. The falling costs of wind energy, coupled with governmental support and favorable policies, are making wind projects increasingly attractive. Furthermore, the industry is becoming a significant source of employment, breathing life into local economies through the construction, operation, and maintenance of wind farms [18].

The wind turbine is coupled to an electrical generator to create direct or alternating current [19]. Most generators used in wind turbines are three-phase asynchronous generators [20]. These have several advantages [21]: they are robust, low-cost and mechanically simple. The downside of the DFIG-based wind turbine is that it is extremely sensitive to load disturbances, variations in the turbine's rotational speed, and variations in the internal and external system parameters [22]. The system is similarly multi-variable, dynamic, tightly linked, and nonlinear, which makes controlling it very challenging. MPPT, which is the primary goal of the DFIG control system, is to efficiently capture wind energy, regardless of the weather [23,24].

Before delving into control strategies, it's essential to grasp the workings of the Doubly Fed Induction Generator. Known for its ability to control both the rotor and stator windings [25], DFIGs offer superior adaptability to varying wind conditions, making them indispensable in the quest for efficient and responsive wind power systems.

### 1.2. Literature review

Numerous studies in existing literature have aimed to develop a control system that is both effective and robust for both the RSC and the GSC of the wind power system's conversion chain to optimize energy production and ensure grid stability. Among these algorithms, linear control laws based on PI controllers have been explored [26], as well as controllers such as DPC or DTC [27], and

non-linear control laws such as SMC [28] and BSC [29].

PI controllers are commonly used in control systems to maintain the stability and performance of dynamic systems [30]. However, this does not preclude the existence of constraints that hinder the achievement of these objectives. The authors [31] apply IFOC based on traditional PI regulators to RSCs and GSCs. However, the major drawback of this control method is the high sensitivity of the control system to the parameters of the system under study and the need to adjust the coefficients of the PI controller. For [32], they propose the improved control method IFOC. This strategy controls the electrical parameters used to implement the regulation of the rotor current and electromagnetic torque components in a DFIG WECS. The disadvantage of this control is that the level of harmonic distortion of the rotor current remains, particularly in the case of low-voltage operating behaviour of the system, commonly known as voltage dips. In this context [33], propose a field-oriented control (FOC) of a rotor DFIG (RT-DFIG) using a high gain observer (HGO). But the observer-based control system may fail in case of current/voltage sensor failure which represents a limitation for HGO control.

Otherwise, in some cases, DPC or DTC controllers may be preferred over PI controllers for wind system control [34]. They offer improved control dynamics, enabling rapid adaptation to changing wind conditions and efficient optimization of energy production [35]. These controllers ensure a rapid response to disturbances, promoting wind system stability in the face of sudden changes in wind conditions or variations in the electrical grid. In addition, their ability to directly adjust electrical power or torque contributes to better power management, improving the efficiency of wind energy conversion [36]. They are also effective at reducing unwanted oscillations, improving the quality of the electricity generated and minimising the impact on the grid [37]. Because of their adaptability to varying conditions, these advanced controllers are well suited to adjusting quickly to changes, ensuring optimum performance in a dynamic wind environment [38]. As in the paper by Ref. [39], the authors considered that flux and torque estimation is the basis of DTC and power estimation is the basis of DPC to keep the bus voltage constant. Also, for the authors [40], they propose an improved control strategy to conventional DPC that limits oscillations under unbalanced grid voltage conditions. Unfortunately, the slow response and large ripples in active and reactive power remain a drawback for this control.

Linear controls often require the linearisation of models to make systems easier to manage. However, linearisation can lead to inaccurate approximations, especially when variations around the nominal operating point are large [41]. For this reason, researchers are turning to non-linear controls that eliminate the need for linearisation. Non-linear controls are often more flexible and adaptive to variations in system parameters [42]. They can handle situations where system parameters change, and are also capable of providing faster responses to changes in conditions or references.

Some non-linear approaches are based on Lyapunov theory, such as SMC [43] and BSC [44,45]. In the paper by Ref. [46], the authors apply a non-linear, super-twist and knock terminal sliding mode control system (ST-FOCSTM) to the two converters RSC and GSC of a WECS, whereas [47] propose the Modified Super Twisting (MST) algorithm for Low Voltage Ride Through (LVRT) improvement under voltage sag. The proposed algorithm is implemented using the second-order sliding mode (SOSM) to control the DFIG wind generator. Nevertheless, they have disadvantages in terms of stability and phenomena known as chattering. Despite their simplicity and robustness in principle, the practical application of SMC is limited by complex online computations and chattering problems related to control switching. Thus, these limitations considerably reduce the control performance expected from such methods in practical implementations.

On the other hand [48], design the non-linear backstepping regulator to control the active and reactive powers of the DFIG with a view to implementing the whole system using the Dspace board, while [49] confirm the robustness of the Backstepping control against parametric variations, by giving a review of the Backstepping control of a wind system based on a DFIG. Despite its design complexity, the Backstepping control algorithm, using the Lyapunov stability technique, provides a stable and robust control that achieves good performance in terms of trajectory tracking and regulation of electrical parameters, thus helping to maximize the energy production of the wind turbine system.

### 1.3. Contribution

Building a solid control system that oversees the proper operation of both the rotor-side and grid-side converters is important to get the most electrical output possible out of a wind power system. The nonlinear hybrid sliding mode Backstepping control law is enhanced in this study, and it is implemented on the DSPACE DS1104 board in order to validate it on a test bench in a lab with a 1.5 kW generator for a wind profile from Al Hoceima region.

The important points that this article contributes can be summed up as follows:

- Present a non-linear model based on DFIG for each component in the conversion chain for wind energy.
- Supply a method for controlling the mechanical rotational speed in order to maximize the power obtained from the wind (MPPT). The reference electromagnetic torque required to create the sliding mode Backstepping control algorithm is fundamentally provided by this.
- Explain the operational principles of the sliding mode and Backstepping control methodologies.
- Develop a robust controller design that uses a new hybrid combination of SM-Backstepping control algorithms applied to the complete wind power conversion chain based on the doubly fed induction generator, to achieve the high performance required in a wind energy system, i.e:
  - o Ensure proper monitoring of electrical reference values such as active and reactive power and DC bus voltage.
  - o Be highly robust to parametric variations, by applying the new hybrid control to the entire wind power system.
  - o To reduce the effect of the “chattering” phenomenon as much as possible.
  - o Reduce the rate of THD and improve the quality of the electrical energy fed into the grid.

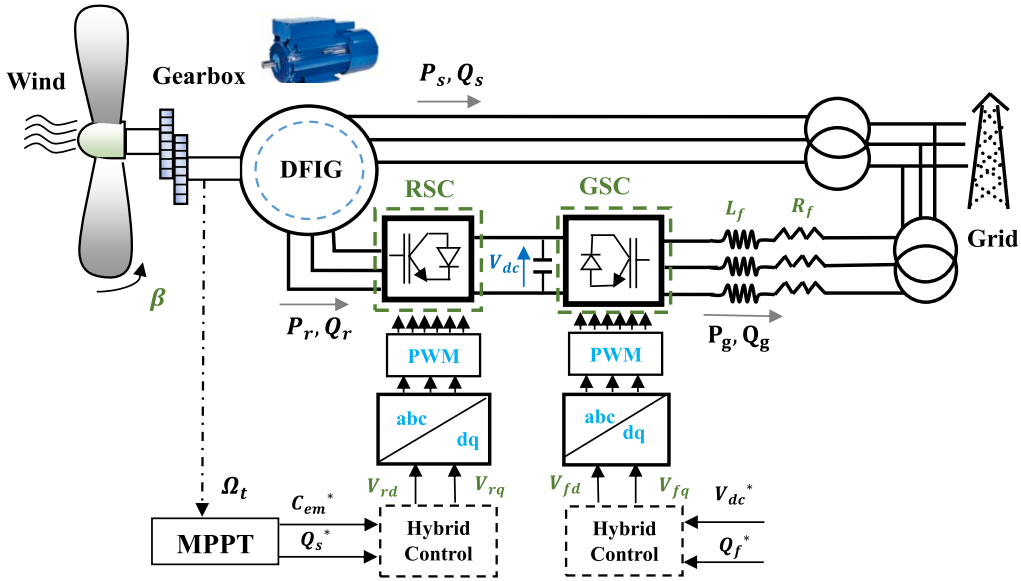


Fig. 1. The components of a wind power system's conversion chain.

- Experimentally validate the hybrid control with Matlab/Simulink software using the real wind profile of Al Hoceima region, and then confirm the effectiveness of the proposed control using a test bench consisting essentially of the DSPACE DS1104 board.

#### 1.4. Organisation of the manuscript

This article is organised as follows: after an introduction, mathematical modelling of each element of the WECS is provided. The principle of sliding mode control is described in detail. Next, the hybrid SM-Backstepping control is well explained and applied to the rotor-side and grid-side converter. Then, the simulation results of the proposed control using the real wind profile of the AL Houceima region are presented. Finally, the whole system is implemented in real time on a test bench using the DSPACE DS1104 board and the ControlDesk in order to visualize the results obtained and verify the correct tracking of the electrical variables and the high robustness of the proposed control algorithm.

## 2. Mathematical modeling of a wind system

As shown in Fig. 1, the wind power system typically consists of a wind turbine, equipped with blades, which captures the kinetic energy of the wind and converts it into mechanical energy. This latter is then transmitted to the DFIG, which transforms the mechanical energy into electrical energy. The system also incorporates energy converters RSC and GSC to convert the generated direct current into alternating current, facilitating its injection into the electrical grid [50].

### 2.1. Wind turbine model

The turbine is modeled by the following equation (1) [51]:

$$P_{aero} = \frac{1}{2} C_p(\lambda, \beta) \cdot \rho \cdot \pi \cdot R^2 \cdot V^3 \quad (1)$$

$\rho$ : Air density (1.225 kg/m<sup>3</sup>)

$$C_p(\lambda, \beta) = 0.5176 \left( \frac{116}{\gamma} - 0.4 * \beta - 5 \right) e^{\frac{-21}{\gamma}} + 0.0068 * \lambda \quad (2)$$

With:

$$\frac{1}{\gamma} = \frac{1}{\lambda + 0.008\beta} - \frac{0.035}{\beta^3 + 1} \quad (3)$$

$$\lambda = \frac{R \cdot \Omega_t}{V} \quad (4)$$

Fig. 2 shows the variation of the power coefficient  $C_p$  with respect to the specified speed  $\lambda$  and pitch angle  $\beta$  by modeling equations

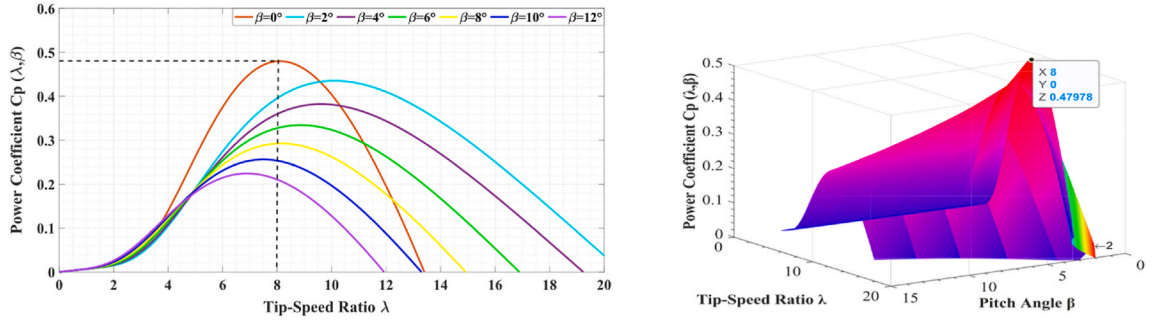


Fig. 2. The power coefficient  $C_p(\lambda, \beta)$ .

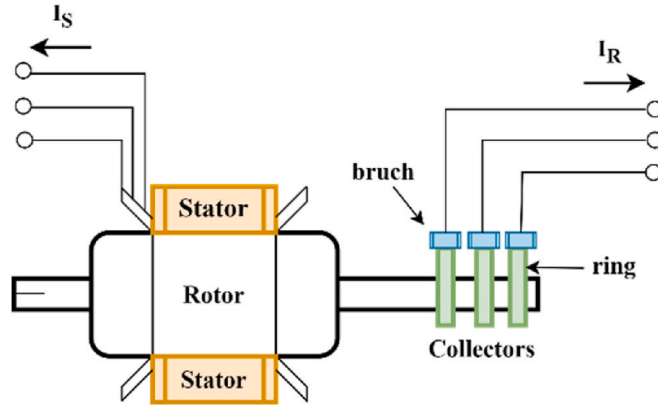


Fig. 3. The structure of DFIG.

(2)–(4) in MATLAB/Simulink. The maximum turbine efficiency for our turbine is achieved if:

$$\beta = 0 ; \lambda_{opt} = 8 ; C_{p-Max} = 0.479$$

Maintaining the optimal values ( $\lambda_{opt}$ ) in the control system for variations in wind conditions is sufficient for achieving MPPT.

The reference electromagnetic torque is then obtained from the estimated torque of the wind turbine torque of the wind turbine (eq (5)):

$$C_{em-ref} = C_g = \frac{C_{t-est}}{G} \tag{5}$$

The electromagnetic reference torque must then be set to the following value (eq (6)):

$$C_{em-ref} = \frac{1}{2} \cdot C_{p-Max} \cdot \rho \cdot \pi \cdot R^5 \cdot V_{est}^3 \cdot \frac{\Omega_{mec}^2}{G^3 \cdot \lambda_{opt}^3} \tag{6}$$

The gearbox is modeled as follows [52] (eq (7)):

$$\begin{cases} \Omega_t = \frac{\Omega_{mec}}{G} \\ C_g = \frac{C_{aero}}{G} \end{cases} \tag{7}$$

The drive shaft transmits the wind turbine’s rotation to the electrical part of the conversion chain. It is modeled by the following mechanical equations (eqs (8) and (9)) [53]:

$$J_{tot} = \frac{J_t}{G^2} + J_g \tag{8}$$

$$C_{mec} = J_{tot} \cdot \frac{d\Omega_{mec}}{dt} = C_g - C_{em} - f \cdot \Omega_{mec} \tag{9}$$

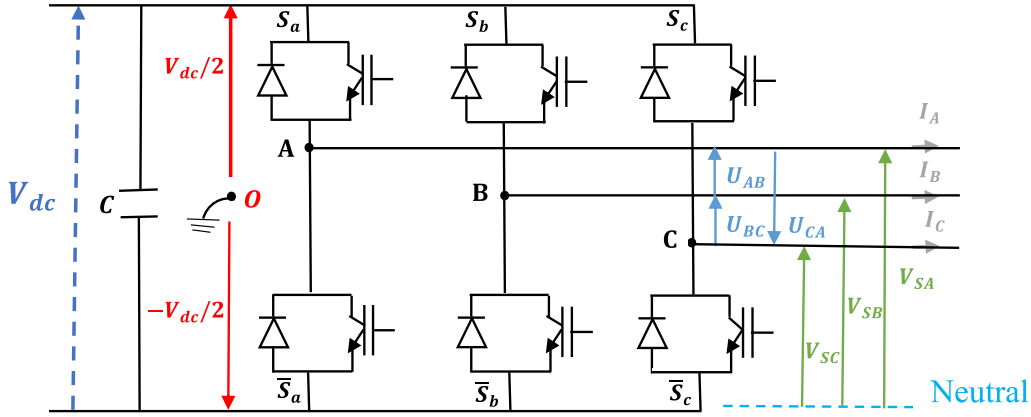


Fig. 4. Converter diagram.

2.2. DFIG model

The DFG consists of a stator, similar to that of a conventional asynchronous machine, and a rotor rotating inside the machine cavity and separated from the stator by an air gap. The three-phase winding of the rotor is accessed by three rings fitted with brushes, as shown in Fig. 3, which allow the rotor voltages to be applied by an external control system [54].

The electrical model of the DFIG in the referential (d,q) is given by:

- Stator and rotor voltage (eqs 10 and 11)

$$\begin{cases} V_{sd} = R_s \cdot I_{sd} + \frac{d\varphi_{sd}}{dt} - \omega_s \cdot \varphi_{sq} \\ V_{sq} = R_s \cdot I_{sq} + \frac{d\varphi_{sq}}{dt} + \omega_s \cdot \varphi_{sd} \\ V_{rd} = R_r \cdot I_{rd} + \frac{d\varphi_{rd}}{dt} - \omega_r \cdot \varphi_{rq} \\ V_{rq} = R_r \cdot I_{rq} + \frac{d\varphi_{rq}}{dt} + \omega_r \cdot \varphi_{rd} \end{cases} \quad (10)$$

With :  $\omega_r = \omega_s - p \cdot \Omega_{mec}$  (11)

-Stator and rotor flux (eqs 12 and 13)

$$\begin{cases} \varphi_{sd} = L_s \cdot I_{sd} + M \cdot I_{rd} \\ \varphi_{sq} = L_s \cdot I_{sq} + M \cdot I_{rq} \\ \varphi_{rd} = L_r \cdot I_{rd} + M \cdot I_{sd} \\ \varphi_{rq} = L_r \cdot I_{rq} + M \cdot I_{sq} \end{cases} \quad (12)$$

$L_s = l_s - M_s, \quad L_r = l_r - M_r$  (13)

-Electromagnetic torque (eq (14))

$G_{em} = p(\varphi_{sd} \cdot I_{sq} - \varphi_{sq} \cdot I_{sd})$  (14)

-Active and Reactive power at stator and rotor (eq (15))

$$\begin{cases} P_s = V_{sd} \cdot I_{sd} + V_{sq} \cdot I_{sq} \\ Q_s = V_{sq} \cdot I_{sd} - V_{sd} \cdot I_{sq} \\ P_r = V_{rd} \cdot I_{rd} + V_{rq} \cdot I_{rq} \\ Q_r = V_{rq} \cdot I_{rd} - V_{rd} \cdot I_{rq} \end{cases} \quad (15)$$

### 2.3. Power converter model

The DFIG is a key component in the conversion chain of a wind power system. It has two windings: a rotor winding connected to the grid via two electronic converters, and a stator winding directly connected to the grid. The two converters can function as an inverter or a rectifier depending on the flow of electrical energy, and their controls allow for the management and optimization of the power generated [55,56].

The structural schema of a three-phase two-level converter is shown in Fig. 4:

All the voltages in Fig. 4 are given by the following equations (16)–(18) [57]:

$$\begin{cases} v_A = \frac{1}{3}(U_{AB} - U_{AC}) = \frac{1}{3}(2 * v_A - v_B - v_C) \\ v_B = \frac{1}{3}(U_{BC} - U_{AB}) = \frac{1}{3}(2 * v_B - v_A - v_C) \\ v_C = \frac{1}{3}(U_{CA} - U_{BC}) = \frac{1}{3}(2 * v_C - v_A - v_B) \end{cases} \quad (16)$$

$$\begin{bmatrix} v_A \\ v_B \\ v_C \end{bmatrix} = \frac{1}{3} \begin{bmatrix} 2 & -1 & -1 \\ -1 & 2 & -1 \\ -1 & -1 & 2 \end{bmatrix} \cdot \begin{bmatrix} v_{A0} \\ v_{B0} \\ v_{C0} \end{bmatrix} * \quad (17)$$

$$\begin{bmatrix} v_{A0} \\ v_{B0} \\ v_{C0} \end{bmatrix} = V_{dc} \cdot \begin{bmatrix} S_a \\ S_b \\ S_c \end{bmatrix} - \frac{V_{dc}}{2} \quad (18)$$

The state of the switches, assumed to be perfect, can be defined by three Boolean control variables  $S_i$  ( $i = a, b, c$ ):

- $S_i = 1$ , the case where the upper switch is closed and the lower switch is open.
- $S_i = 0$ , the case where the top switch is open and the bottom switch is closed.

Under these conditions, simple voltages can be written in the following matrix form (eq (19)):

$$\begin{bmatrix} v_A \\ v_B \\ v_C \end{bmatrix} = \frac{V_{dc}}{3} \begin{bmatrix} 2 & -1 & -1 \\ -1 & 2 & -1 \\ -1 & -1 & 2 \end{bmatrix} \cdot \begin{bmatrix} S_a \\ S_b \\ S_c \end{bmatrix} \quad (19)$$

The use of this type of converter makes it possible to obtain Pulse Width Modulation output signal shapes whose modularity makes it possible to limit interference by modifying the frequency spectrum of the signal (rejection of the first non-zero harmonics towards high frequencies).

### 2.4. DC bus model

The time evolution of the DC bus voltage is obtained from the following integration (eq (20)) [58]:

$$\begin{cases} W_{dc} = \int P_c \cdot dt = \frac{1}{2} \cdot C \cdot V_{dc}^2 \\ \frac{dV_{dc}^2}{dt} = \frac{2}{C} (P_f - P_r) \end{cases} \quad (20)$$

### 2.5. RL filter model

The voltage equations modeling the filter are written in PARK's reference frame (d,q) as follows (eq (21)) [59]:

$$\begin{cases} v_{fd} = -R_f \cdot I_{fd} - L_f \cdot \frac{dI_{fd}}{dt} + \omega_s \cdot L_f \cdot I_{fq} \\ v_{fq} = -R_f \cdot I_{fq} - L_f \cdot \frac{dI_{fq}}{dt} - \omega_s \cdot L_f \cdot I_{fd} + v_s \end{cases} \quad (21)$$

## 3. Theory of sliding mode control

The central idea behind SMC is to create a sliding surface, a hyperplane in state space, along which the system's state trajectory should ideally "slide". The sliding surface is defined by an equation or set of equations that relate the system's states. SMC passes through three stages [60]:

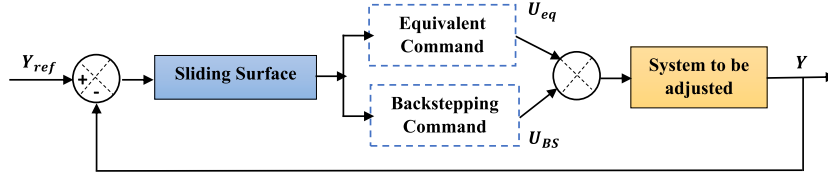


Fig. 5. The sliding-backstepping mode Controller's composition.

- The choice of the sliding surface (eq (22))

$$S(x) = \left( \frac{d}{dt} + \delta \right)^{r-1} \cdot e(x) \quad (22)$$

- The establishment of conditions of existence

In order to ensure both the stability of the non-linear system and the attraction of the variable to be regulated towards its reference value, LYAPUNOV proposes a  $V(x)$  function (eqs (23) and (24)) [61]:

$$V(x) = \frac{1}{2} \cdot S(x)^2 \quad (23)$$

$$\dot{V}(x) = S(x) \cdot \dot{S}(x) < 0 \quad (24)$$

- The control law's determination

Two terms make up the sliding mode control (eq (25)) [62]:

$$u = u_{eq} + u_n \quad (25)$$

$u_{eq}$ : this vector describes an ideal slip movement independent of the uncertainties and disturbances generated by the system. The following invariance conditions for the slip surface give its expression (eq (26)):

$$\begin{cases} \dot{S}(x) = 0 \\ S(x) = 0 \end{cases} \quad (26)$$

For discontinuous control, the simplest of those found in the literature is as follows (eq (27)) [63]:

$$u_n = K \cdot \text{sign}(S(x)) \quad (27)$$

$K$ : the control gain

The sign function returns  $-1, 0$  or  $+1$  when the numeric expression value is negative, zero or positive.

Although sliding mode control has several advantages, it also has a disadvantage known as "chattering", which can lead to unwanted vibrations due to rapid and frequent variations in control around the sliding surface [64].

#### 4. Hybrid sliding-backstepping mode control

In this paper, a new technique is developed to overcome the chattering drawback of SMC in the pursuit of higher performance. Hybrid sliding mode backstepping is the term used to describe this method as shown in Fig. 5.

SMC and BSC are the two control approaches on which the hybrid sliding-backstepping control theory is built. By guiding the system along a predefined sliding surface, SMC tries to make the system insensitive to shocks and uncertainty. The difference between the system's actual output and the desired reference output is used to define this sliding surface. The goal is to use a suitable control law to force the system to stay on this sliding surface. As a result, the system is resistant to errors and changes in the system [65].

BSC, on the other hand, is a control technique that involves designing a series of control laws by working "backwards" through the system dynamics. Each step in the backstepping process adds a control term that cancels out previous errors and leads the system towards a desired reference trajectory. The aim is to design sequential control laws that stabilize the system and lead it towards precise trajectory-following performance [66].

In hybrid Sliding-Backstepping control, sliding mode control is used to maintain the system in a finite-slip subspace. When the system reaches this space, backstepping control is activated to guide the system to the desired reference trajectory using specific control laws.

By combining these two techniques, hybrid Sliding-Backstepping control offers both robustness to disturbances and uncertainties and high trajectory-following accuracy.



## 5. Rotor side converter control

- Determining the equivalent control  $V_{rdeq}$  and  $V_{rreq}$

Applying SMC to the DFIG (eqs (28) and (29)) [67]:

$$\begin{cases} e_1 = P_{sref} - P_s \\ e_2 = Q_{sref} - Q_s \end{cases} \quad (28)$$

$$\begin{cases} \dot{e}_1 = \dot{P}_{sref} - \dot{P}_s \\ \dot{e}_2 = \dot{Q}_{sref} - \dot{Q}_s \end{cases} \quad (29)$$

With (eq (30) and (31)):

$$\begin{cases} \dot{P}_s = -\frac{v_s \cdot M}{L_s} i_{rq} \\ \dot{Q}_s = \frac{v_s^2}{\omega_s \cdot L_s} - \frac{v_s \cdot M}{L_s} i_{rd} \\ \dot{i}_{rd} = \frac{V_{rd}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} i_{rd} + \omega_r \cdot i_{rq} \\ \dot{i}_{rq} = \frac{V_{rq}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} i_{rq} - \omega_r \cdot i_{rd} - \omega_r \cdot \frac{M \cdot V_s}{L_r \cdot L_s \cdot \sigma \cdot \omega_s} \end{cases} \quad (30)$$

$$\begin{cases} \dot{e}_1 = \dot{P}_{sref} + \frac{V_s \cdot M}{L_s} \left( \frac{V_{rq}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} i_{rq} - \omega_r \cdot i_{rd} - \omega_r \cdot \frac{M \cdot V_s}{L_r \cdot L_s \cdot \sigma \cdot \omega_s} \right) \\ \dot{e}_2 = \dot{Q}_{sref} + \frac{V_s \cdot M}{L_s} \left( \frac{V_{rd}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} i_{rd} + \omega_r \cdot i_{rq} \right) \end{cases} \quad (31)$$

As already mentioned, the sliding mode controller consists of two terms (eq (32)):

$$U = U_{eq} + U_n \quad (32)$$

As a result, the following gives the expression for the derivative of the sliding surface (eq (33)):

$$\begin{cases} \dot{e}_1 = \dot{P}_{sref} + \frac{V_s \cdot M}{L_s} \left( \frac{V_{rreq} + V_{rqn}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} i_{rq} - \omega_r \cdot i_{rd} - \omega_r \cdot \frac{M \cdot V_s}{L_r \cdot L_s \cdot \sigma \cdot \omega_s} \right) \\ \dot{e}_2 = \dot{Q}_{sref} + \frac{V_s \cdot M}{L_s} \left( \frac{V_{rdeq} + V_{rdn}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} i_{rd} + \omega_r \cdot i_{rq} \right) \end{cases} \quad (33)$$

During sliding mode and permanent mode, we have (eq (34)):

$$\begin{cases} e_{1,2} = 0 \\ \dot{e}_{1,2} = 0 \\ V_{rdn} = V_{rqn} = 0 \end{cases} \quad (34)$$

The expression for the derivative of the sliding surface becomes (eq (35)):

$$\begin{cases} V_{rdeq} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \cdot \dot{Q}_{sref} + R_r \cdot i_{rd} - \omega_r \cdot L_r \cdot \sigma \cdot i_{rq} \\ V_{rreq} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \cdot \dot{P}_{sref} + R_r \cdot i_{rq} + \omega_r \cdot L_r \cdot \sigma \cdot i_{rd} + \omega_r \cdot V_s \cdot \frac{M}{L_s \cdot \omega_s} \end{cases} \quad (35)$$

- Determining the discontinuous control  $V_{rdn}$  and  $V_{rqn}$

The RSC is subjected to the backstepping control:

Calculating reference rotor currents is the first step.

The following equation gives the Lyapunov function connected to the stator's active and reactive power errors (eq (36)) [68]:

$$V_1 = \frac{1}{2} \cdot e_1^2 + \frac{1}{2} \cdot e_2^2 \quad (36)$$

Its derivative is stated as follows:

$$\dot{V}_1 = e_1 \cdot \dot{e}_1 + e_2 \cdot \dot{e}_2 = e_1 \left[ \dot{P}_{sref} + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} \left( V_{rq} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right] + e_2 \left[ \dot{Q}_{sref} + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} (V_{rd} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq}) \right] \quad (37)$$

According to Lyapunov,  $\dot{V}_1$  must be negative in order to guarantee the stability of the subsystem. For this, we select it in the following format:

$$\dot{V}_1 = -k_1 \cdot e_1^2 - k_2 \cdot e_2^2 \leq 0 \quad (38)$$

When equations (37) and (38) are equivalent, the following result is obtained (eq (39)):

$$e_1 \left[ \dot{P}_{sref} + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} \left( V_{rq} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right] + e_2 \left[ \dot{Q}_{sref} + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} (V_{rd} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq}) \right] = -k_1 \cdot e_1^2 - k_2 \cdot e_2^2 \quad (39)$$

Which give: (eq (40))

$$\begin{cases} e_1 \left[ \dot{P}_{sref} + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} \left( V_{rq} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right] = -k_1 \cdot e_1^2 \\ e_2 \left[ \dot{Q}_{sref} + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} (V_{rd} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq}) \right] = -k_2 \cdot e_2^2 \end{cases} \quad (40)$$

The following defines the expression of the virtual control  $I_{rd}$  and  $I_{rq}$  (eq (41)):

$$\begin{cases} I_{rqref} = \left[ \frac{L_r \cdot \sigma \cdot L_s}{R_r \cdot V_s \cdot M} \left( \dot{P}_{sref} + k_1 \cdot e_1 + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} \left( V_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right) \right] \\ I_{rdref} = \frac{L_r \cdot \sigma \cdot L_s}{R_r \cdot V_s \cdot M} \left[ \dot{Q}_{sref} + k_2 \cdot e_2 + \frac{V_s \cdot M}{L_r \cdot \sigma \cdot L_s} (V_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq}) \right] \end{cases} \quad (41)$$

Calculating the rotor voltages in step two.

We shall derive the real control laws  $V_{rqn}$  and  $V_{rdn}$  that allow the system's design goals to be fulfilled.

The rotor current errors are determined according to the criteria below (eq (42)) [69]:

$$\begin{cases} e_3 = I_{rqref} - I_{rq} \\ e_4 = I_{rdref} - I_{rd} \end{cases} \quad (42)$$

Their derivatives are given by (eq (43)):

$$\begin{cases} \dot{e}_3 = \dot{I}_{rqref} - \dot{I}_{rq} \\ \dot{e}_4 = \dot{I}_{rdref} - \dot{I}_{rd} \end{cases} \quad (43)$$

So (eq (44)):

$$\begin{cases} \dot{e}_3 = \dot{I}_{rqref} - \frac{1}{L_r \cdot \sigma} \left( V_{rqn} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \\ \dot{e}_4 = \dot{I}_{rdref} - \frac{1}{L_r \cdot \sigma} \left( V_{rdn} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq} \right) \end{cases} \quad (44)$$

The expanded Lyapunov function takes on the following form (eq (45)):

$$V_2 = \frac{1}{2} \cdot e_1^2 + \frac{1}{2} \cdot e_2^2 + \frac{1}{2} \cdot e_3^2 + \frac{1}{2} \cdot e_4^2 \quad (45)$$

The derivative of it is as follows (eq (46)):

$$\dot{V}_2 = e_1 \cdot \dot{e}_1 + e_2 \cdot \dot{e}_2 + e_3 \cdot \dot{e}_3 + e_4 \cdot \dot{e}_4 \quad (46)$$

Which give (eq (47)):

$$\dot{V}_2 = \dot{V}_1 + e_3 \cdot \dot{e}_3 + e_4 \cdot \dot{e}_4 \quad (47)$$

$$\dot{V}_2 = \dot{V}_1 + e_3 \left[ \dot{I}_{rqref} - \frac{1}{L_r \cdot \sigma} \left( V_{rqn} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right] + e_4 \left[ \dot{I}_{rdref} - \frac{1}{L_r \cdot \sigma} (V_{rdn} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq}) \right] \quad (48)$$

For the system to be stable,  $\dot{V}_2$  must be negative. In order to do this, we select  $\dot{V}_2$  in the form [70]:

$$\dot{V}_2 = -k_1 \cdot e_1^2 - k_2 \cdot e_2^2 - k_3 \cdot e_3^2 - k_4 \cdot e_4^2 = \dot{V}_1 - k_3 \cdot e_3^2 - k_4 \cdot e_4^2 \leq 0 \quad (49)$$

By equating these two equations (48) and (49), we obtain (eq (50)):

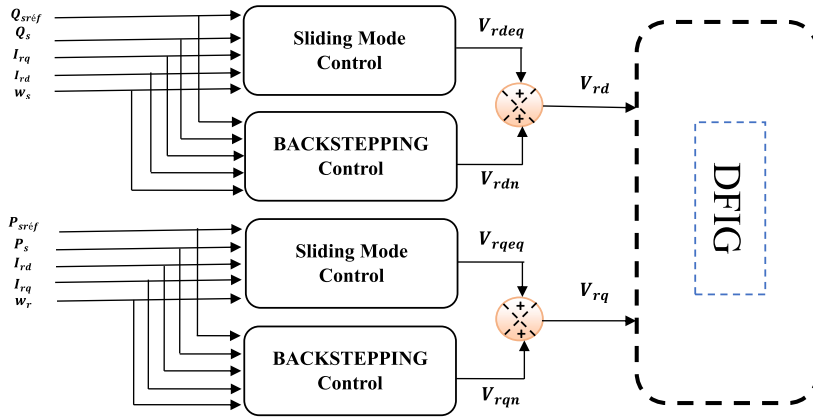


Fig. 6. Application of hybrid control on the RSC.

$$\dot{V}_1 - k_3 \cdot e_3^2 - k_4 \cdot e_4^2 = \dot{V}_1 + e_3 \left[ \dot{I}_{rqref} - \frac{1}{L_r \cdot \sigma} \left( V_{rqn} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right] + e_4 \left[ \dot{I}_{rdref} - \frac{1}{L_r \cdot \sigma} \left( V_{rdn} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq} \right) \right] \quad (50)$$

So (eq (51)):

$$\begin{cases} e_3 \left[ \dot{I}_{rqref} - \frac{1}{L_r \cdot \sigma} \left( V_{rqn} - R_r \cdot I_{rq} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} - g \cdot \frac{M \cdot V_s}{L_s} \right) \right] = -k_3 \cdot e_3^2 \\ e_4 \left[ \dot{I}_{rdref} - \frac{1}{L_r \cdot \sigma} \left( V_{rdn} - R_r \cdot I_{rd} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rq} \right) \right] = -k_4 \cdot e_4^2 \end{cases} \quad (51)$$

This provides the formulation of the true global control,  $V_{rdn}$  and  $V_{rqn}$  defined by (eq (52)):

$$\begin{cases} V_{rdn} = L_r \cdot \sigma \left[ k_4 \cdot e_4 + \dot{I}_{rdref} + \frac{1}{L_r \cdot \sigma} \left( R_r \cdot I_{rd} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rq} \right) \right] \\ V_{rqn} = L_r \cdot \sigma \left[ k_3 \cdot e_3 + \dot{I}_{rqref} + \frac{1}{L_r \cdot \sigma} \left( R_r \cdot I_{rq} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} + g \cdot \frac{M \cdot V_s}{L_s} \right) \right] \end{cases} \quad (52)$$

With:  $k_3, k_4$  are positive constants.

**- Global Control generation using Sliding-Backstepping Mode control**

The first term " $u_{eq}$ " produced by the sliding-mode control and the second term " $u_n$ " produced by the backstepping control constitute the hybrid sliding-backstepping mode controller for the two rotor voltage vectors  $V_{rd}$  and  $V_{rq}$  as shown in Fig. 6 and (eqs (53) and (54)).

$$u_{MG-BS} = u_{eq-MG} + u_{n-BS} \quad (53)$$

$$\begin{cases} V_{rd} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \cdot \dot{Q}_{sref} + R_r \cdot I_{rd} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rq} + L_r \cdot \sigma \left[ k_4 \cdot e_4 + \dot{I}_{rdref} + \frac{1}{L_r \cdot \sigma} \left( R_r \cdot I_{rd} - L_r \cdot \sigma \cdot \omega_r \cdot I_{rq} \right) \right] \\ V_{rq} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \cdot \dot{P}_{sref} + R_r \cdot I_{rq} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} + \omega_r \cdot \frac{M \cdot V_s}{L_s \cdot \omega_s} + L_r \cdot \sigma \left[ k_3 \cdot e_3 + \dot{I}_{rqref} + \frac{1}{L_r \cdot \sigma} \left( R_r \cdot I_{rq} + L_r \cdot \sigma \cdot \omega_r \cdot I_{rd} + g \cdot \frac{M \cdot V_s}{L_s} \right) \right] \end{cases} \quad (54)$$

**6. Grid side converter control**

-Determining the equivalent control  $V_{fdcq}$  and  $V_{fqeq}$

Applying sliding-mode control to the DFIG for  $r = 1$ , the active and reactive power slip area is given by Ref. [71] (eq (55)):

$$\begin{cases} e_5 = P_{fref} - P_f \\ e_6 = Q_{fref} - Q_f \end{cases} \quad (55)$$

The errors' derivatives are as follows (eq (56)):

$$\begin{cases} \dot{e}_5 = \dot{P}_{fref} - \dot{P}_f \\ \dot{e}_6 = \dot{Q}_{fref} - \dot{Q}_f \end{cases} \quad (56)$$

With (eqs (57) and (58)):

$$\begin{cases} P_f = v_s \cdot I_{fq} \\ Q_f = -v_s \cdot I_{fd} \\ \dot{I}_{fq} = -\frac{v_{fq}}{L_f} - \frac{R_f}{L_f} \cdot I_{fq} - \omega_s \cdot I_{fd} + \frac{v_s}{L_f} \\ \dot{I}_{fd} = -\frac{v_{fd}}{L_f} - \frac{R_f}{L_f} \cdot I_{fd} + \omega_s \cdot I_{fq} \end{cases} \quad (57)$$

$$\begin{cases} \dot{e}_5 = \dot{P}_{fref} - v_s \cdot \left( -\frac{v_{fq}}{L_f} - \frac{R_f}{L_f} \cdot I_{fq} - \omega_s \cdot I_{fd} + \frac{v_s}{L_f} \right) \\ \dot{e}_6 = \dot{Q}_{fref} + v_s \cdot \left( -\frac{v_{fd}}{L_f} - \frac{R_f}{L_f} \cdot I_{fd} + \omega_s \cdot I_{fq} \right) \end{cases} \quad (58)$$

The derivative of the sliding surface becomes (eq (59)):

$$\begin{cases} \dot{e}_5 = \dot{P}_{fref} + \frac{v_s \cdot R_f}{L_f} \cdot I_{fq} + \frac{v_s (v_{fqeq} + v_{fqn})}{L_f} + v_s \cdot \omega_s \cdot I_{fd} - \frac{v_s^2}{L_f} \\ \dot{e}_6 = \dot{Q}_{fref} - \frac{v_s \cdot R_f}{L_f} \cdot I_{fd} - \frac{v_s (v_{fd eq} + v_{fdn})}{L_f} + v_s \cdot \omega_s \cdot I_{fq} \end{cases} \quad (59)$$

In sliding and permanent mode, we get the equivalent control expressed by (eq (60)):

$$\begin{cases} V_{fd eq} = \frac{L_f}{V_s} \cdot \dot{Q}_{sref} - R_f \cdot I_{fd} + L_f \cdot \omega_s \cdot I_{fq} \\ V_{fq eq} = \frac{L_f}{V_s} \cdot \dot{P}_{sref} - R_f \cdot I_{fq} - L_f \cdot \omega_s \cdot I_{fd} + V_s \end{cases} \quad (60)$$

-Determining the discontinuous control  $V_{fdn}$  and  $V_{fqn}$

The GSC is subjected to the Backstepping control:

Calculate the  $I_{fd}$  and  $I_{fq}$  filter currents in step one

The following equation describes the Lyapunov function linked to the filter's active and reactive power errors (eq (61)):

$$V_3 = \frac{1}{2} \cdot e_5^2 + \frac{1}{2} \cdot e_6^2 \quad (61)$$

The derivative of it is as follows:

$$\dot{V}_3 = e_5 \cdot \dot{e}_5 + e_6 \cdot \dot{e}_6 = e_5 \left( \dot{P}_{fref} - v_s \cdot \left( -\frac{v_{fqn}}{L_f} - \frac{R_f}{L_f} \cdot I_{fq} - \omega_s \cdot I_{fd} + \frac{v_s}{L_f} \right) \right) + e_6 \left( \dot{Q}_{fref} + v_s \cdot \left( -\frac{v_{fdn}}{L_f} - \frac{R_f}{L_f} \cdot I_{fd} + \omega_s \cdot I_{fq} \right) \right) \quad (62)$$

For the system to be stable,  $\dot{V}_3$  must be negative. In order to do this, we select  $\dot{V}_3$  in the form [72]:

$$\dot{V}_3 = -k_5 \cdot e_5^2 - k_6 \cdot e_6^2 \leq 0 \quad (63)$$

With:  $k_5, k_6$  are positive constants

By equating these two equations (62) and (63), we get (eq (64)):

$$e_5 \left( \dot{P}_{fref} - v_s \cdot \left( -\frac{v_{fqn}}{L_f} - \frac{R_f}{L_f} \cdot I_{fq} - \omega_s \cdot I_{fd} + \frac{v_s}{L_f} \right) \right) + e_6 \left( \dot{Q}_{fref} + v_s \cdot \left( -\frac{v_{fdn}}{L_f} - \frac{R_f}{L_f} \cdot I_{fd} + \omega_s \cdot I_{fq} \right) \right) = -k_5 \cdot e_5^2 - k_6 \cdot e_6^2 \quad (64)$$

Which give (eq (65)):

$$\begin{cases} e_5 \left( \dot{P}_{fref} - v_s \cdot \left( -\frac{v_{fqn}}{L_f} - \frac{R_f}{L_f} \cdot I_{fq} - \omega_s \cdot I_{fd} + \frac{v_s}{L_f} \right) \right) = -k_5 \cdot e_5^2 \\ e_6 \left( \dot{Q}_{fref} + v_s \cdot \left( -\frac{v_{fdn}}{L_f} - \frac{R_f}{L_f} \cdot I_{fd} + \omega_s \cdot I_{fq} \right) \right) = -k_6 \cdot e_6^2 \end{cases} \quad (65)$$

The following defines the expression of the virtual control  $I_{fd}$  and  $I_{fq}$  (eq (66)):

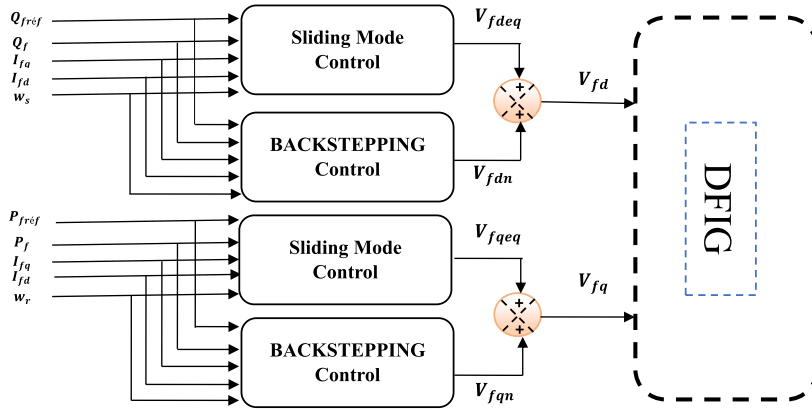


Fig. 7. Application of hybrid control on the GSC.

$$\begin{cases} I_{fdref} = -\frac{L_f}{R_f \cdot v_s} \left( \dot{P}_{fref} + v_s \left( \frac{V_{fqn}}{L_f} + \omega_s \cdot I_{fd} - \frac{v_s}{L_f} \right) + k_5 \cdot e_5 \right) \\ I_{fqref} = \frac{L_f}{R_f \cdot v_s} \left( \dot{Q}_{fref} + v_s \cdot \left( -\frac{V_{fdn}}{L_f} + \omega_s \cdot I_{fq} \right) + k_6 \cdot e_6 \right) \end{cases} \quad (66)$$

The upcoming system's preferred reference will be this control.

Calculating the filter voltages in step two  $V_{fdn}$  and  $V_{fqn}$

To accomplish the design goals for the entire system, we shall deduce the genuine control laws  $V_{fdn}$  and  $V_{fqn}$  in this stage (eq (67)).

$$\begin{cases} e_7 = I_{fdref} - I_{fd} \\ e_8 = I_{fqref} - I_{fq} \end{cases} \quad (67)$$

The derivatives of the errors are provided by (eqs (68) and (69)):

$$\begin{cases} \dot{e}_7 = \dot{I}_{fdref} - \dot{I}_{fd} \\ \dot{e}_8 = \dot{I}_{fqref} - \dot{I}_{fq} \end{cases} \quad (68)$$

$$\begin{cases} \dot{e}_7 = \dot{I}_{fdref} + \frac{v_{fdn}}{L_f} + \frac{R_f}{L_f} \cdot I_{fd} - \omega_s \cdot I_{fq} \\ \dot{e}_8 = \dot{I}_{fqref} + \frac{v_{fqn}}{L_f} + \frac{R_f}{L_f} \cdot I_{fq} + \omega_s \cdot I_{fd} - \frac{v_s}{L_f} \end{cases} \quad (69)$$

The "LYAPUNOV" function is given as follows (eq (70)):

$$V_4 = \frac{1}{2} \cdot e_7^2 + \frac{1}{2} \cdot e_8^2 \quad (70)$$

The derivative of it is as follows:

$$\dot{V}_4 = e_7 \cdot \dot{e}_7 + e_8 \cdot \dot{e}_8 = e_7 \left( \dot{I}_{fdref} + \frac{v_{fdn}}{L_f} + \frac{R_f}{L_f} \cdot I_{fd} - \omega_s \cdot I_{fq} \right) + e_8 \left( \dot{I}_{fqref} + \frac{v_{fqn}}{L_f} + \frac{R_f}{L_f} \cdot I_{fq} + \omega_s \cdot I_{fd} - \frac{v_s}{L_f} \right) \quad (71)$$

According to LYAPUNOV, in order to guarantee the stability of the system, a negative function,  $\dot{V}_4$ , must be used. We select  $\dot{V}_4$  in the following format for this [73]:

$$\dot{V}_4 = -k_7 \cdot e_7^2 - k_8 \cdot e_8^2 \leq 0 \quad (72)$$

With:  $k_7, k_8$  are positive constants.

By equating these two equations (71) and (72), we obtain:

$$e_7 \left( \dot{I}_{fdref} + \frac{v_{fdn}}{L_f} + \frac{R_f}{L_f} \cdot I_{fd} - \omega_s \cdot I_{fq} \right) + e_8 \left( \dot{I}_{fqref} + \frac{v_{fqn}}{L_f} + \frac{R_f}{L_f} \cdot I_{fq} + \omega_s \cdot I_{fd} - \frac{v_s}{L_f} \right) = -k_7 \cdot e_7^2 - k_8 \cdot e_8^2 \quad (73)$$

Which give (eq (74)):

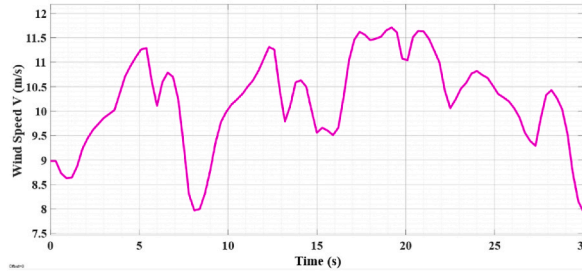


Fig. 8. Al Hoceima Morocco's variable wind speed profile.

Table 1  
Abbreviations.

MPPT	Maximum Power Point Tracking	BSC	Backstepping Control
DFIG	Doubly Fed Induction Generator	DPC	Direct Power Control
RSC	Rotor Side Converter	DTC	Direct Torque Control
GSC	Grid Side Converter	PI	Proportional Integral
FOC	Field Oriented Control	WECS	Wind Energy Conversion System
IFOC	Indirect Field Oriented Control	THD	Total Harmonic Distortion
SMC	Sliding Mode Control	PWM	Pulse Width Modulation

$$\begin{cases} e_7 \left( \dot{I}_{fdref} + \frac{v_{fdn}}{L_f} + \frac{R_f}{L_f} I_{fd} - \omega_s I_{fq} \right) = -k_7 \cdot e_7^2 \\ e_8 \left( \dot{I}_{fqref} + \frac{v_{fqm}}{L_f} + \frac{R_f}{L_f} I_{fq} + \omega_s I_{fd} - \frac{v_s}{L_f} \right) = -k_8 \cdot e_8^2 \end{cases} \quad (74)$$

The expression of the global real control  $V_{fdn}$  and  $V_{fqm}$  defined by (eq (75)):

$$\begin{cases} v_{fdn} = -L_f \left( k_7 \cdot e_7 + \dot{I}_{fdref} + \frac{R_f}{L_f} I_{fd} + \omega_s I_{fq} \right) \\ v_{fqm} = -L_f \left( k_8 \cdot e_8 + \dot{I}_{fqref} + \frac{R_f}{L_f} I_{fq} + \omega_s I_{fd} - \frac{v_s}{L_f} \right) \end{cases} \quad (75)$$

### - Global Control Generation through Sliding-Backstepping Mode Control

According to Fig. 7, the hybrid sliding-backstepping mode controller for the two filter voltage vectors  $V_{fd}$  and  $V_{fq}$  is made up of the first term " $u_{eq}$ " produced by the sliding-mode control and the second term " $u_n$ " produced by the Backstepping control (eqs (76) and (77)).

$$u_{MG-BS} = u_{eq-MG} + u_{n-BS} \quad (76)$$

So:

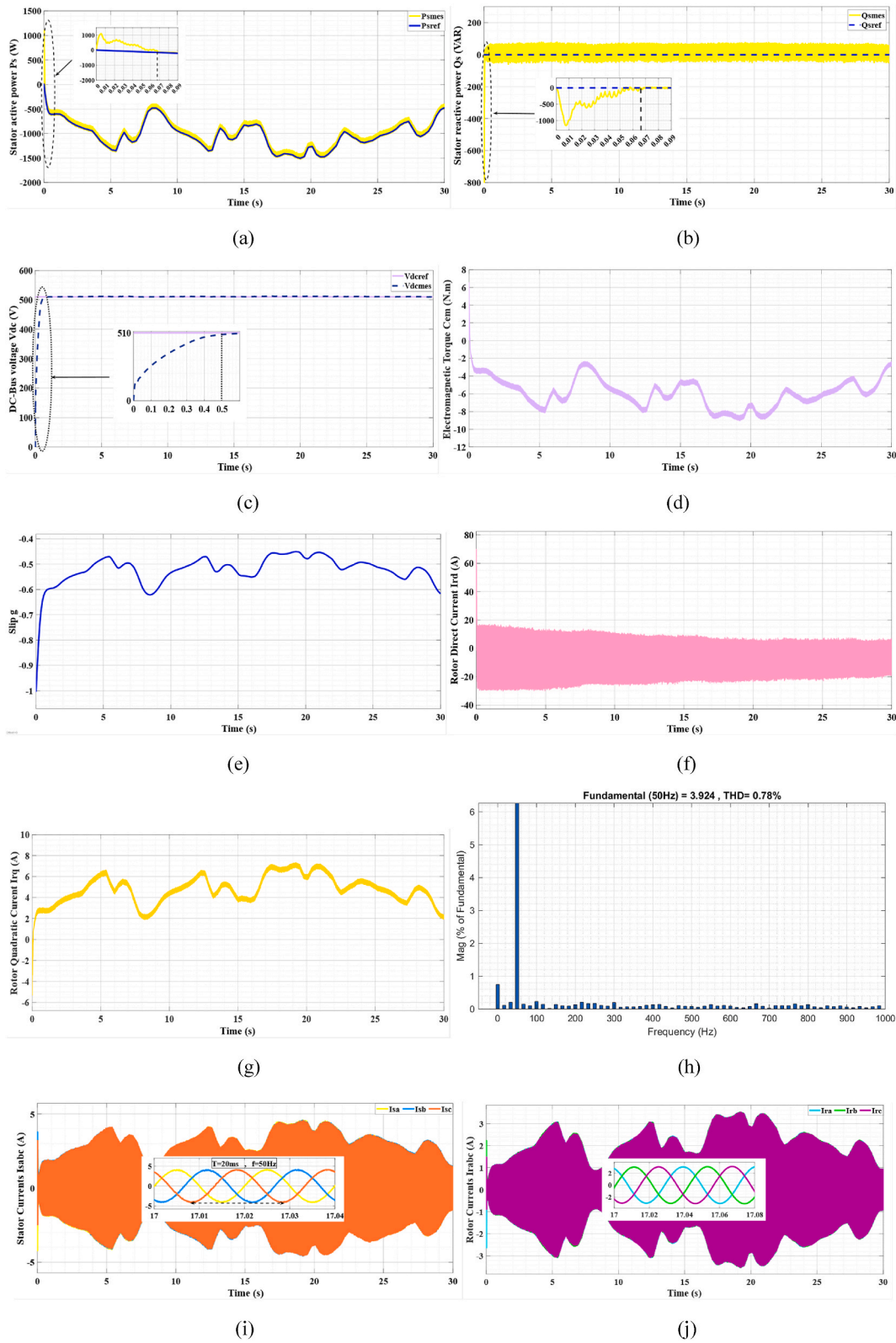
$$\begin{cases} v_{fd} = \frac{L_f}{V_s} \cdot \dot{Q}_{sref} - R_f \cdot I_{fd} + L_f \cdot I_{fq} \cdot \omega_s - L_f \left( k_5 \cdot e_5 + \dot{I}_{fdref} + \frac{R_f}{L_f} I_{fd} + \omega_s I_{fq} \right) \\ v_{fq} = \frac{L_f}{V_s} \cdot \dot{P}_{sref} - R_f \cdot I_{fq} - L_f \cdot I_{fd} \cdot \omega_s + v_s - L_f \left( k_6 \cdot e_6 + \dot{I}_{fqref} + \frac{R_f}{L_f} I_{fq} + \omega_s I_{fd} - \frac{v_s}{L_f} \right) \end{cases} \quad (77)$$

## 7. Results of simulation

In order to test and validate the control developed above, two tests were carried out:

The first test concerns the monitoring and regulation of the wind power system, taking the aerodynamic power according to the MPPT strategy as the reference for the active power of the DFIG stator and zero as the reference for the reactive power of the stator, in order to guarantee a unitary power factor on the stator side and therefore optimize the quality of the energy returned to the grid.

The second test aims at evaluating the robustness of the control with respect to parametric variations. In this section, a real wind profile from AL Hoceima region on the north coast of Morocco is shown in Fig. 8, applied to a wind system based on a 1.5 kW DFIG (Table 1). Simulation results for the two tests are given in Figs. 9 and 10 respectively.



**Fig. 9.** Simulation results for hybrid control of a wind power system based on DFIG: (a) Active Power  $P_s$ , (b) Reactive power  $Q_s$ , (c) DC Bus voltage  $V_{dc}$ , (d) Electromagnetic Torque  $C_{em}$ , (e) Slip  $g$ , (f) Direct current  $I_{rd}$ , (g) Quadratic current  $I_{rq}$ , (h) Stator currents THD, (i) Stator currents  $I_{sabc}$ , (j) rotor currents  $i_{rbrc}$ .

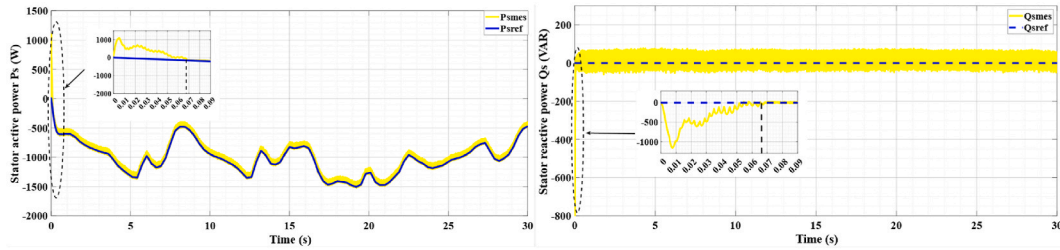


Fig. 10. Robustness test with variation of resistances  $R_r$  and  $R_s$  for variable wind speed.

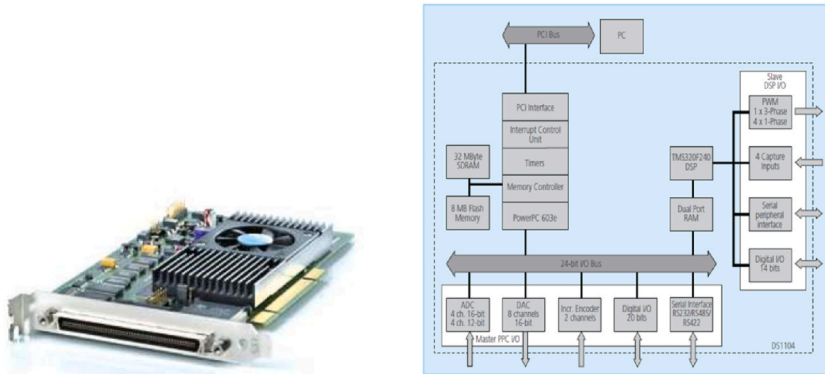


Fig. 11. Illustration of the DS1104 Board and DS1104 architecture.

**- Pursuit Tests for Variable speed response**

The active power has the same inverse shape as the wind profile, and the fact that the machine is operating in generator mode explains the negative sign of the active power. It follows its reference from time  $t = 0.069s$  as shown in Fig. 9 (a) with almost no deviation. While the reactive power follows its zero reference with a response time of  $0.069s$  as shown in Fig. 9 (b) with a small error.

With regard to the VDC response, Fig. 9 (c) shows that the DC bus voltage requires  $0.5s$  to achieve the constant reference value of  $510V$ . While Fig. 9 (d) shows the electromagnetic torque varying inversely with the wind profile. Fig. 9 (e) also shows that operation in hyper-synchronous mode is assured and proved by the negative sign of the slip.

The direct rotor current varies close to a small value since it depends on the reactive power, as shown in Fig. 9 (f). The rotor quadrature current has the inverse form of the active power, as indicated in Fig. 9 (g).

During wind variation, the stator and rotor currents maintain a sinusoidal shape with a period of  $0.02s$  as shown in Fig. 9 (i) and (j) respectively with a frequency equal to the mains frequency of  $50Hz$ .

The normalised THD value must be less than 5% for satisfactory power quality. In this study, the THD was less than 0.78%, as shown in Fig. 9 (h), indicating good power quality.

**- Robustness tests for Variable speed response**

The preceding test was conducted with the machine’s characteristics fixed; nevertheless, these parameters are subject to a variety of physical phenomena, including temperature, which can cause resistance values to increase, inductor saturation, etc. Due to measurement tools and the used methodology, the identification of these factors is also stated in infidelities. To achieve this, the  $R_r$  and  $R_s$  values are increased by 50% compared with their nominal values. Fig. 10 shows the results obtained.

Fig. 10 shows that even if resistor and inductance value have changed, active and reactive power remain in line with reference values, and also remain decoupled. This shows that hybrid control by sliding mode backstepping is highly robust to parametric variations in the DFIG, due to good monitoring of power setpoints, with an almost identical response time on start-up.

**8. Experimental validation**

**-Presentation of the experimental banc.**

The hardware includes an FPGA Nexys3 board, a Semikron IGBT inverter, a Danfoss variable speed drive, a 1.5 KW squirrel cage induction machine (SCIM), a 1.5 KW DFIG, and a DSPACE 1104. Following the creation of the wind emulator, the inverter is used to apply the control algorithm to the DFIG rotor. The test bench summary is depicted in Fig. 12, and the actual realized test bench in the



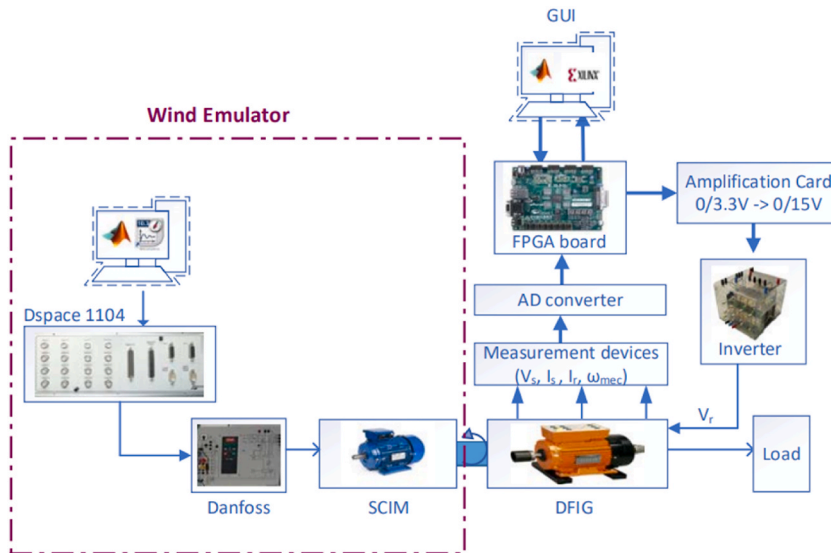


Fig. 12. Synopsis of the test bench.

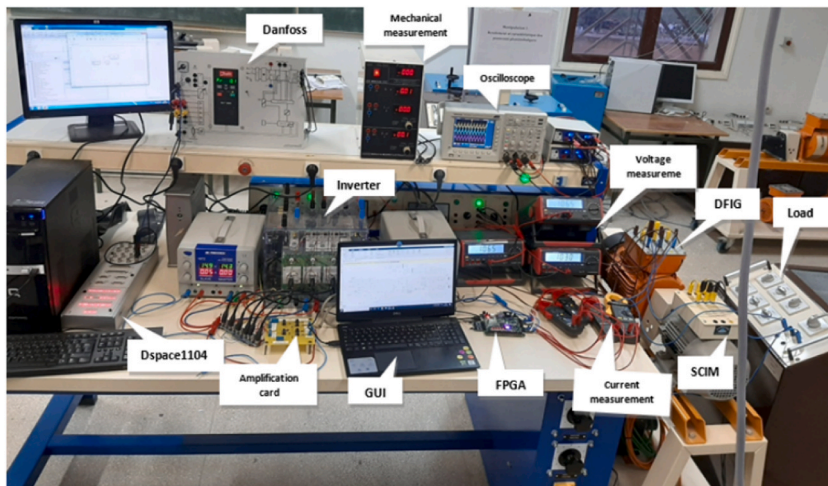


Fig. 13. Test bench.

laboratory is shown in Fig. 13.

**- Wind Emulator**

The DS1104 Board, as shown in Fig. 11 is used in the construction of the wind emulator. It’s an interface board that can translate Simulink blocks into DSP-based system machine code. It comprises 8 ADC analog-to-digital converters with input voltages from  $-10\text{ V}$  to  $+10\text{ V}$ , and 8 DAC digital-to-analog converters with output voltages from  $-10\text{ V}$  to  $+10\text{ V}$ . It also features several interfaces, including digital inputs and outputs, incremental encoders, etc. The DS1104 board also features a slave DSP, the TMS320F240 DSP, which will be used to generate PWM signals.

The development of the control system using Simulink blocks, system simulation, and real-time model execution via the Ds1104 board are the first steps in the prototype process. Through the use of the Dspace1104 board and the Danfoss drive (variable speed drive), the emulator is implemented. Next, this drive instructs the DFIG and SCIM to work together to achieve a mechanical speed that matches the applied wind profile.

**- Stability analysis**

After modeling and simulating the hybrid sliding-mode backstepping control of a DFIG-based wind power system on Matlab/

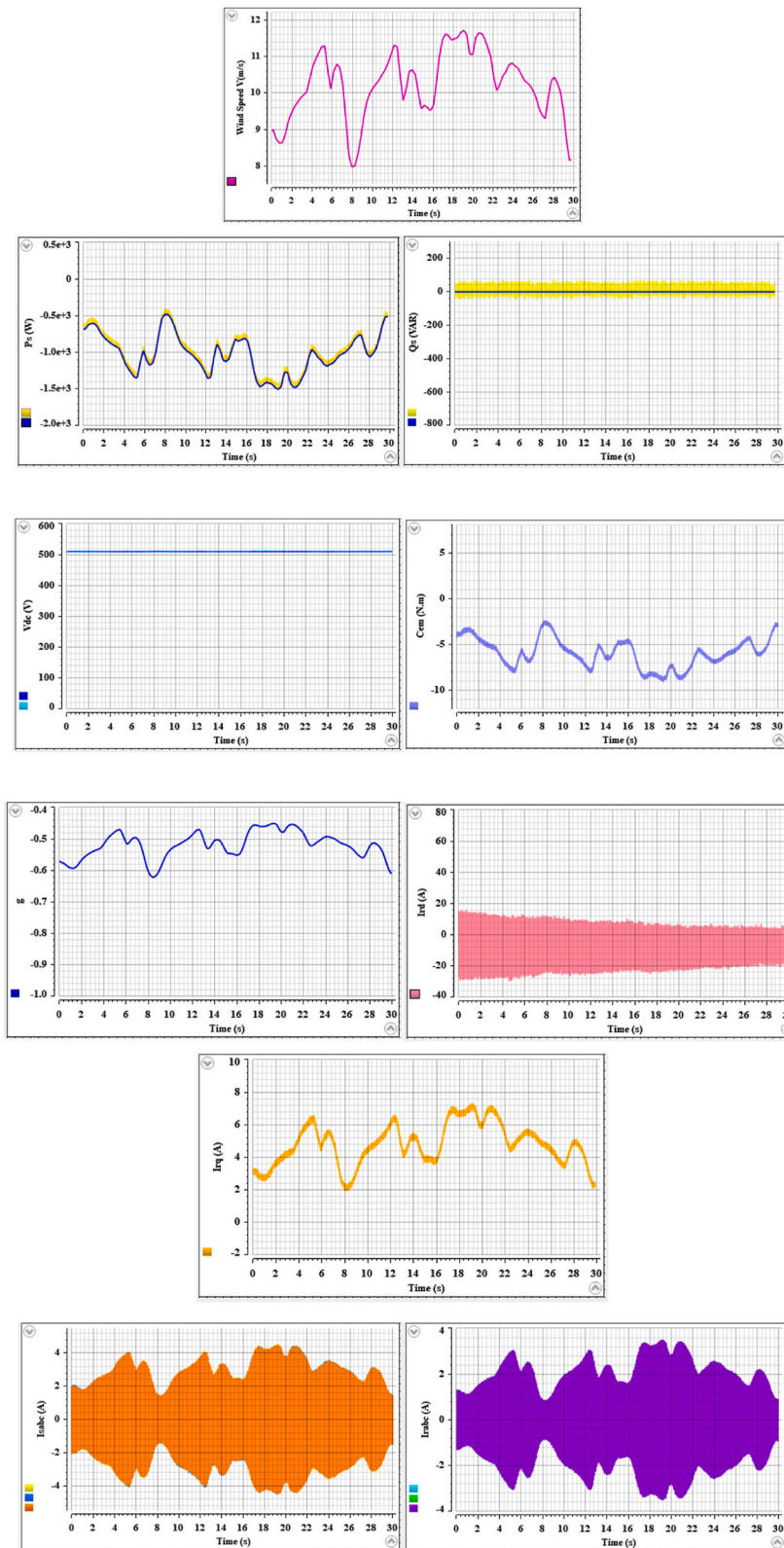


Fig. 14. DSPACE Control-Desk real time simulation results.

**Table 2**

Comparison between our proposed method and other controller technologies published recently.

Ref	Control method	Response Time (s)	Error (%)	THD (%)	Robustness
[65]	FOC	0.8	0.3184	–	Low
	Backstepping adaptive	0.1	1.47	–	
[42]	Sliding Mode	0.0858	–	3.99	Medium
	Backstepping	0.09	–	2.36	High
[35]	DPC based SM	–	–	5.56	Low
<i>Proposed technique</i>	Hybrid SM-BS	0.069	0.2	0.78	High

Simulink, real-time control is carried out using Al Hoceima region's profile and a computer connected to the Dspace DS1104 board with an RTI interface to Simulink through which they can convert Simulink blocks into machine code that can be run on a DSP-based system and also using ControlDesk that is used to display in real time the different variables of the program developed in Simulink. Next Generation 5.4 version 2.0.

The DS1104 Board, as shown in Fig. 11, comprises 8 ADC analog-to-digital converters with input voltages from  $-10$  V to  $+10$  V, and 8 DAC digital-to-analog converters with output voltages from  $-10$  V to  $+10$  V. It also features several interfaces, including digital inputs and outputs, incremental encoders, etc. The DS1104 board also features a slave DSP, the TMS320F240 DSP, which will be used to generate PWM signals.

ControlDesk is an interface that allows you to visualize in real time various variables in the program developed in Simulink, and also to modify parameters defining the operating mode of the blocks making up the Simulink diagram.

The experimental findings for a DFIG-based wind power system are shown in Fig. 14 after loading the sdf file on Control Desk from the Matlab/Simulink model at the RTI interface. This was done by applying hybrid sliding-mode Backstepping control to both the RSC and GSC converters. With regard to these findings, we observe improved grid power quality based on shape (sinusoidal with a frequency of 50 Hz) and decreased THD. In comparison to references, the active and reactive powers provide superior tracking. In terms of speed, THD accuracy, and robustness, the experimental findings are remarkably similar to the simulation results.

## - COMPARATIVE STUDY

Table 2 presents a comparison study of the results between the control developed and another recent research. The reduction in response time for the first two controls [65], FOC and Adaptive Backstepping, is undoubtedly the main benefit of the hybrid SM-Backstepping control. It's also noteworthy that this study's THD was significantly lower than any of the five suggested control groups. In conclusion, in terms of performance, robustness, and quality, the SM-Backstepping hybrid controller provides the higher power generation.

## 9. Conclusion

The blending of sliding-mode and backstepping control has resulted in the hybrid control highlighted in this article. It used them to bring together the advantages of each and finally obtain a control that is efficient and robust against parametric variations. Starting with a study of all the elements in the conversion chain of the DFIG-based wind power system, followed by a clear insight into the principle of the sliding mode control law strategy and Backstepping, as well as the hybrid control mechanism and their applications to the two converters RSC and GSC.

Finally, accuracy and robustness against parametric variations are verified by simulation on Matlab/Simulink software and by real-time implementation on the DSPACE DS1104 board in a test bench. When compared to previous control systems, the SM-Backstepping hybrid controller has demonstrated significant potential in terms of quality and tracking reference since it has been able to eradicate the "CHATTERING" issue without compromising system performance. When compared to other current techniques, all of the results obtained in terms of tracking, robustness, and quality of electrical energy provided to the grid satisfy the needed performance criteria, indicating the effectiveness of the suggested controller. In the future, other hybridizations will be considered to achieve the desired performance in the case of an unbalanced grid.

## CRedit authorship contribution statement

**Farah Echiheb:** Writing – original draft, Formal analysis, Data curation, Conceptualization. **Ismail Elkafazi:** Software, Methodology, Investigation. **Badre Bossoufi:** Validation, Supervision, Software. **Brahim El bhiri:** Visualization, Supervision, Project administration. **Mishari Metab Almalki:** Validation, Software, Resources. **Thamer A.H.Alghamdi:** Software, Resources, Project administration, Funding acquisition.

## Declaration of competing interest

The authors declare the following financial interests/personal relationships which may be considered as potential competing interests: Badre BOSSOUFI reports financial support was provided by Sidi Mohamed Ben Abdellah University Faculty of Sciences Dhar El

Mahraz. Badre BOSSOUFI reports a relationship with Sidi Mohamed Ben Abdellah University Faculty of Sciences Dhar El Mahraz that includes: employment. If there are other authors, they declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

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